

NAVAL POSTGRADUATE SCHOOL Monterey, California



THESIS

A STUDY OF SPECTRUM LOADING AND RANGE-PAIR COUNTING METHOD EFFECTS ON CUMULATIVE FATIGUE DAMAGE

by

John Scott Atkinson, Jr.

March 1977

Thesis Advisor:

G. H. Lindsey

Approved for public release; distribution unlimited.

DDC MNV 19 1977

AD NO.

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered) READ INSTRUCTIONS BEFORE COMPLETING FORM REPORT DOCUMENTATION PAGE REPORT NUMBER RECIPIENT'S CATALOG NUMBER 2. GOVT ACCESSION NO. Master's Thesis, TITLE (and Subtitle) A Study of Spectrum Loading and Range-March 1977 Pair Counting Method Effects on Cumu-6. PERFORMING ORG. REPORT NUMBER lative Fatigue Damage . AUTHOR(e) S. CONTRACT OR GRANT NUMBER(s) John Scott/Atkinson, Jr PERFORMING ORGANIZATION NAME AND ADDRESS 10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS Naval Postgraduate School Monterey, California 93940 1. CONTROLLING OFFICE NAME AND ADDRESS 12. REPORT DATE Marca 1977 Naval Postgraduate School 13. NUMBER OF PAGES Monterey, California 93940 117 4. MONITORING AGENCY NAME & ADDRESS(II dillorent from Controlling Office) 18. SECURITY CLASS. (of this report) Unclassified Naval Postgraduate School Sa. DECLASSIFICATION/DOWNGRADING Monterey, California 93940 16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited. 17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different A IS. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) fatigue damage computer program range-pair counting method spectrum loading sequence loading eros side if necessary and identify by block member) This thesis is a study of cumulative fatigue damage. Variations in cumulative fatigue damage resulting from block loading spectra and randomized cycle loading spectra are investigated. Fatigue damage results show the merit of counting load cycles using the range-pair counting method. Complete FORTRAN computer program documentation enables this thesis to serve as a program user's manual.

DD 1 JAN 73 1473 (Page 1)

EDITION OF 1 NOV 65 IS OBSOLETE 5/N 0102-014-6601 | 251 450

SECURITY CLASSIFICATION OF THIS PAGE (Show Date Entered

SECURITY CLASSIFICATION OF	THIS PAGETWHEN Det	a Entered.	
	i		

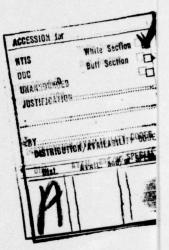
Approved for public release; distribution unlimited.

A Study of Spectrum Loading and Range-Pair Counting Method Effects on Cumulative Fatigue Damage

by

John Scott Atkinson, Jr. Lieutenant, United States Navy B.S., Randolph-Macon College, 1968

Submitted in partial fulfillment of the requirements for the degree of



MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL March 1977

Author

Approved by:

Thesis Advisor

Chairman, Department of Aeronautics

A 1. Tomm

Dean of Science and Engineering

ABSTRACT

This thesis is a study of cumulative fatigue damage.

Variations in cumulative fatigue damage resulting from block loading spectra and randomized cycle loading spectra are investigated. Fatigue damage results show the merit of counting load cycles using the range-pair counting method.

Complete FORTRAN computer program documentation enables this thesis to serve as a program user's manual.

TABLE OF CONTENTS

I.	INTRODUCTION	13
II.	GENERAL DESCRIPTION OF THE COMPUTER PROGRAM	15
	A. INPUT DATA REQUIREMENTS	18
	B. RANGE-PAIR COUNTING METHOD	22
	C. DAMAGE CALCULATION	26
III.	DISCUSSION OF RESULTS	29
IV.	RECOMMENDATIONS FOR FUTURE RESEARCH	39
COMPU'	TER OUTPUT	40
COMPU	TER PROGRAM	93
LIST (OF REFERENCES	116
INITI	AL DISTRIBUTION LIST	117

LIST OF TABLES

ı.	FREQUENCY OF LOAD LEVELS FROM MILSPEC A	16
II.	FATIGUE DAMAGE DUE TO BLOCK LOADING	30
III.	FATIGUE DAMAGE DUE TO RANDOMIZED CYCLE LOADING USING RANGE-PAIR COUNTING METHOD	30
IV.	FATIGUE DAMAGE DUE TO RANDOMIZED CYCLE LOADING NOT USING RANGE-PAIR COUNTING METHOD	31

LIST OF DRAWINGS

1.	ILLUSTRATION OF RANGE-PAIR COUNTING METHOD	25
2.	PLOT SHOWING ELASTIC-PERFECTLY PLASTIC BEHAVIOR -	27
3.	COFFIN-MANSON DIAGRAM	27
4.	PLOT OF RANDOMIZED FLIGHT LOADS	33
5.	PLOT OF RANDOMIZED FLIGHT LOADS	34
6.	PLOT OF RANDOMIZED FLIGHT LOADS	35
7.	PLOT OF RANDOMIZED FLIGHT LOADS	36
8.	PLOT OF RANDOMIZED FLIGHT LOADS	37
9.	PLOT OF RANDOMIZED FLIGHT LOADS	38

LIST OF SYMBOLS

A Coefficient of the x^2 term in the equation of a line on a constant life fatigue diagram where minimum₂ stress is x and maximum stress is y. $(R = Ax^2 + Bx + C - y)$

AA An assigned value of +1. or -1.

AAA A stress used in the calculation of plastic strain.

ABDIF The absolute value of DIF.

ABM The absolute value of ASMAX or of ASMIN, as assigned.

ABMAX The absolute value of ASMAX.

ABMEAN The absolute value of ASMEAN

ABMIN The absolute value of ASMIN.

ABR4 The absolute value of R(4).

ABR7 The absolute value of R(7).

ABS The name of a routine calling for the absolute value of a quantity.

AKT Stress concentration factor, K₊.

ASMAX The product (AKT) (STMAX).

ASMEAN The quantity (ASMAX + ASMIN)/2.

ASMIN The product (AKT) (STMIN).

AVSGMN Average value of SIGMIN over an interval.

AVSGMX Average value of SIGMAX over an interval.

B Coefficient of the x term. (see A)

BBB A stress used in the calculation of plastic strain.

C The constant. (see A)

COFMAN Inverse of the Coffin-Manson slope.

CYCINT The number of cycles in an interval.

CYCLES The calculated number of cycles expected to be indicated on a constant life fatigue diagram for the applied combination of maximum and minimum stress.

C1 The Residual Stress Relaxation Constant. (see ENEP)

DAM Damage.

DECK Decimal or real value of integer K after conversion.

DEL2 A portion of a least-squares-method solution.

DIF The difference between residual stress and equilibrium residual stress. (RES(I) - EQRES)

DO2 A portion of a least-squares-method solution.

DUMMY A variable used in the calculation of the number of cycles to be considered as an interval for relaxation determination.

ELMOD The elastic modulus.

EN The number of cycles from the beginning of the relaxation process to the end of the current interval.

ENEP The number of cycles required for overload residual stress effect to return to within one-tenth of its original difference from equilibrium conditions.

 $(N_{ep} = C1/(ABM)^{E1} (ABMEAN)^{E2})$

ENN The number of applied cycles at a load level.

ENNCYC The ratio of the number of applied cycles to the number of cycles to failure. (ENN/CYCLES)

EPSD LCF strain intercept.

EQRES Equilibrium residual stress.

EX An exponential function depicting the relaxation of residual stress.

EXP The name of a routine calling for the exponential value of a quantity.

EXPO An exponent. The power of 10 which indicates the number of cycles to failure.

El Residual stress relaxation exponent.

E2 Residual stress relaxation exponent.

FLOAT The name of a routine calling for integer-to-real conversion.

I A variable subscript.

IBLOCK The identifying number of a block, the blocks being numbered consecutively from 1 to NBLOCK.

IFIX The name of a routine calling for real-to-integer conversion.

IN The number of steps input to the range-pair counting subroutine.

IPRINT Value controlling the WRITE statements.

IRAIN A counter.

IRPCM Value controlling entry into the range-pair counting subroutine.

ISTEP The identifying step number, the steps being numbered from 1 to NLEVEL.

ITYPE The identifying type number, the types being numbered from 1 to NTYPE.

J A variable subscript.

JA Value of +1 or 0, as assigned for branch determination.

JB Value of -1 or 0, as assigned for branch determination.

JJ An index variable.

JJJ An index variable.

JKL An index variable.

K An index variable.

KK An index variable.

KPMAX The number of steps output from the range-pair counting subroutine.

L An index variable.

LMN An index variable.

M An index variable.

N An index variable with values of N=4-7 indicating the power of 10, and thus identifying a particular life cycle curve.

NBLOCK The total number of times to execute a block of loads.

NDECK The number of data decks to be run sequentially.

NFLAG An integer used as a counter.

NFLAG2 An integer used as a counter.

NLEVEL The total number of steps, or levels, of loads in a block.

NN A subscripted variable used to indicate which types of loads are experienced in which blocks.

NTYPE The total number of different types.

PLSTRA Plastic strain.

R Residue term in damage calculation.

RES Residual stress.

RNCYC The number of cycles for a level after exitting the range-pair counting subroutine.

SIGMAX Maximum stress.

SIGMIN Minimum stress.

STMAX Maximum applied stress.

STMIN Minimum applied stress.

SUMDEL Summation of damage for a flight.

SUMENN Accumulated total of applied cycles. (Summation of ENN)

SUMNC Accumulated cycle ratio. (Summation of ENN/CYCLES)

SUMR Summation of R(N), N=4,7.

SUMRN Summation of NR(N), N=4,7.

SUMR2 Summation of $R(N)^2$, N=4,7.

SUMR2N Summation of $NR(N)^2$, N=4,7.

SUMR3 Summation of $R(N)^3$, N=4,7.

SUMR4 Summation of $R(N)^4$, N=4,7.

TITLE1 Identification of the source of the S-N data.

TLL Tensile limit load.

TM1, TM2 Material type.

TTYS One-fifth of tensile yield stress.

TYS Tensile yield stress.

T1,T2 Test identifying information.

X Variable equivalent to SIGMIN.

Y Variable equivalent to SIGMAX.

I. INTRODUCTION

In an aircraft fatigue life monitoring program, determination of the actual loading environment which produces the fatigue is a major problem. With both military trainer and fighter aircraft, which may be used for a variety of duties, regular or continuous load recording programs have to be considered mandatory. Such load recording programs provide the data to calculate the consumed fatigue life of operating aircraft and are used in selecting design spectra for future aircraft designs.

To facilitate the calculations of fatigue life from such data, it is important to know not only the loading magnitudes but also the time sequence of the loading. The fatigue life of a local critical point varies due to residual stresses remaining after the application of a load causing local plasticity in tension or compression. For instance, a peak tensile load into the plastic flow region leaves a residual compressive stress, which lowers the local magnitude of the following tensile stresses, thus increasing the fatigue life. In a similar manner a peak compressive load into the plastic flow region, if it were to occur, would leave a residual tensile stress, which decreases the fatigue life.

Currently, fatigue monitoring of naval aircraft is based on the total number of g readings recorded at four

Selected levels by an exceedence level counting accelerometer. Using micropressors in the near future it will be possible to record each maximum and minimum load level experienced by an aircraft in sequence. The data collected can be used to monitor the fatigue life of a structure via the determination of damage accumulated at a point found to be critical in a structural test of a prototype. It should be noted that this calculation uses a theoretical model.

Damage is not observable or measurable.

The objective of this thesis is to use a computer program, employing the range-pair counting technique on time sequence recorded maximum and minimum aircraft loads and a relaxation model to consider residual stresses, to calculate plastic, elastic, and total aircraft structural damage.

II. GENERAL DESCRIPTION OF THE COMPUTER PROGRAM

The computer program used in this fatigue analysis is divided into four modules. Each module is clearly labeled in the program listing (page 93). In module I the input data are read and assimilated in preparation for future calculations. The input parameter requirements are presented in detail beginning on page 18, and six sample sets of input data are illustrated starting on page 40.

When the load sequence randomization technique is used to input loading data, it is a part of module I. The load sequence randomization technique uses the computer library subroutine RANDU to place 10 percent of the MIL spectrum A positive loads (Table I) in a random order. Each of the positive loads is paired with a minimum load of 11 percent limit load, or 1-g. During the randomization each load has an equal probability of selection. A counter restricts the number of times a value is selected to the number of occurrences of the particular load level in MIL spectrum A.

Two variations on the randomization of the MIL spectrum A load levels are available. The first option permits randomization of the negative MIL spectrum A load levels using the same technique as in the case of the positive MIL spectrum A load levels. Each negative load level is randomly paired with a positive load level. Since there are not as many negative load levels in MIL spectrum A as

TABLE I

FREQUENCY OF MANEUVER LOADS

NUMBER OF TIMES PER THOUSAND HOURS THAT LOAD FACTOR IS EXPERIENCED

FLIGHT MANEUVER LOAD SPECTRUM A		
17000		
9500		
6500		
4500		
2500		
1360		
440		
150		
40		
16		
TOTAL 42006		
FLIGHT MANEUVER LOAD SPECTRUM A		
500		
200		
100		
60		
35		
30		
25		
20 15		
10 5		
3		
TOTAL 1003		

there are positive load levels, the excess load levels are paired with 1-g loads. Mixing of the negative MIL spectrum A load levels and the 1-g load levels is accomplished using the computer library subroutine RANDU. For this ordering there is an 80 percent chance that each negative load selected will be a 1-g load. A counter ensures the proper numbers of each load type are included. Eighty percent probability of selection of a 1-g load was used to spread the smaller number of negative MIL spectrum A loads throughout the sequence, and also to speed computer operation by not having the matrix of negative MIL spectrum A loads addressed so often, after the counter indicated the matrix elements had all been used.

The second option for randomizing loads provides for the inclusion of a ground cycle between each flight. The ground cycle loading is taken as -2500 psi, or -8 percent limit load. Eight percent is the level of the ground cycle load used by the A7 manufacturer. There are 4201 positive loading events in 100 hours of MIL spectrum A flight time. Each flight is arbitrarily taken as one hour in length, so a ground load is paired with every forty-second positive loading event.

In module II the local stresses and strains are determined from the input data. If the local peak stress is in the plastic region, the specimen is assumed to unload elastically, leaving a local residual compressive stress in the material. In the analysis developed by Potter [Ref. 6],

the transient portion of the residual stress so produced relaxes toward zero, or an equilibrium residual stress. After the stresses and strains are calculated, the local stress cycles are counted in module III using the range-pair counting technique. In module IV the fatigue damage is calculated. Damage is determined separately for elastic and plastic strain events. The elastic and plastic damage for each flight is listed as output, as well as the total cumulative damage from any previous flights.

A. INPUT DATA REQUIREMENTS (MODULE I)

Samples of data are listed starting on page 40; formatting is described beginning on page 93. Both pages should be consulted while reading the following descriptions.

1. Data Card 1

Only one of these cards is required or permitted for each program run.

NDECK = the number of data decks to be run sequentially. The input on this card will be common to all data decks run. It is not necessary for the different data decks to have any parameters in common.

IPRINT = the value controlling the write statements.

Output available during the process of the analysis

includes:

a. Maximum and minimum applied stress of each cycle and the local stress response throughout the spectrum. Also printed out is the residual stress, equilibrium stress, applied cycles, and the equilibrium period for each cycle.

- b. The elastic local stress history as input into the range-pair subroutine and the resulting range-paired spectrum.
- c. The maximum plastic strain occurrence during the spectrum and the damage associated with each strain reversal.
- d. The accumulated damage associated with the plastic strains.
- e. The range-paired elastic stress spectrum and the damage associated with each level for block loading, or for each cycle for RANDU generated loadings.
- f. The accumulated damage associated with the current block of loading, including the plastic strain damage, and the total damage since the initiation of cycling.

The value assigned to IPRINT controls which pieces of data are included in the printout.

If IPRINT = 1, all six items listed above are printed for each flight or block of loads.

If IPRINT = 2, all items listed above except b,
are-printed.

If IPRINT = 3, only items d and f listed above are
printed.

An example of each output option is given beginning on page . When the number of loads is more than fifty, the output is quite voluminous, and a paper usage default may interrupt program operation. With large numbers of loads, IPRINT should be set equal to 3.

IRPCM = the value controlling entry into the rangepair counting subroutine.

If IRPCM = 1, the range-pair counting method is used.

If IRPCM = 2, the range-pair counting method is
skipped.

Each data deck run sequentially must contain the following cards. The number of data decks was indicated by the value assigned NDECK.

2. Card 1

Information on this card is descriptive, alphanumeric data used only as a heading test identification.

3. Card 2

The name of the material type and four material constants are listed on this card. The material type is an alphanumeric entry. The tensile yield stress, LCF strain intercept, inverse of the Coffin-Manson slope, and the modulus of elasticity are real number entries. The LCF strain intercept is the numerical value of the ordinate intercept in the log-log Coffin-Manson plot of plastic strain range vs. cycles to failure. Units of tensile yield stress and modulus of elasticity are ksi.

4. Cards 3 Through 6

Each card contains the three coefficients of a second order least-squares curve fit for S-N data obtained from a Goodman diagram published in MIL-HDBK-5A. The curves are fit for lives of 10^4 , 10^5 , 10^6 , and 10^7 cycles. Space is provided on each card for an alphanumeric indication of

the data source. Only the data source printed on card 6 will be retained in computer storage for subsequent printout. Columns 72 through 80 are used for information only and are not read by the computer.

5. Card 7

Constants to be used in the equilibrium period calculations are listed on this card.

6. Card 8

The stress concentration factor is the only entry on this card.

7. Card 9

The first entry on this card is the number of blocks of loads, or the number of flights. The terms block and flight are used interchangeably. This entry is the number of times the list of loads is to be repeated. The next entry is the number of loads. The number of loads will always be the same as the quantity of cards 11 used as input. The last entry is the number of types of loads. Loads are indicated as being a different type in order to alter the loading pattern during subsequent flights. It is important to understand that the program will always consider load types in ascending numerical order. A flight listing load types in the order 1, 2, 3, 4 will produce identical results with a flight listing load types in the order, 4, 3, 2, 1.

8. Card 10

Only the limit load is listed here, with units of ksi. The maxima and minima of all cycles are input as a decimal fraction of this load entry.

9. Cards 11

Information identifying each load cycle is presented on these cards. The first entry is an integer representing the step, or number of the load. This integer increments by one on each card, with the last card having a value equal to NLEVEL. Other entries on these cards are the type of load, the minimum and maximum decimal fraction of the limit load for the particular step, and the number of consecutive cycles at this load level. It is important to notice that the format requires the number of cycles to be entered as a real number, although fractions of cycles will not be processed.

10. Cards NLEVEL + 11

The first entry is an integer representing the number of the block, or flight. This entry is a dummy variable and is not used in any subsequent calculations. The other entries on this card indicate which types of loads are to be processed on this particular flight. Each type of load may be processed only once on each flight. The order in which the types are listed is of no significance, since the load-types are automatically processed in ascending numerical order.

B. RANGE-PAIR COUNTING METHOD

In assessing the life consumed by an individual aircraft, damage is calculated from Miner's Law using S-N data. This type of life calculation is of greater value if the cycle counting method used takes into account the actual load-time

history of the aircraft. Many load cycle counting methods have been developed and used. Fatigue test experience indicates that a useful counting technique for aircraft fatigue must take into account the loading sequence as well as the magnitude of the maximum and minimum load peaks [Ref. 10]. The two most realistic counting techniques are the rangepair counting method and the rain-flow counting method. Both methods experimentally yield approximately the same results. The major difference between the two methods is that the rain-flow method counts in terms of load ranges, or half cycles, whereas the range-pair method records the time history in terms of complete load cycles. When counting in terms of full cycles, the two methods are equivalent. The merit of counting half cycles is not important when analyzing aircraft fatigue because the large number of load reversals does not permit a half cycle to influence the numerical results. The range-pair counting method corresponds to the stable cyclic stress-strain behavior of a material in that strain ranges counted as cycles will form closed stress-strain hysterisis loops [Ref. 4].

The computer program employed in this thesis has the option of using the range-pair counting method, or simply using each peak in the order that it occurs. When the range-pair counting method is not used, only momentary load values are considered, and information regarding the cyclic stress-strain pattern, which is important in fatigue calculations, is lost. Also, minor load variations, which are of little

importance in fatigue calculations, are counted as additional cycles.

The range-pair counting method is illustrated in Figure 1. It counts a strain range as a cycle if it can be paired with a subsequent straining of equal magnitude in the opposite direction. For a complicated load history, some of the ranges counted as cycles will be simple ranges, such as 2-3, during which the strain does not change direction, but others, such as 1-8, will be interrupted by smaller ranges which will also be counted as cycles. In Figure 1 ranges are marked with solid lines and the paired ranges with dashed lines.

Each peak is taken in order as the initial peak of a range, except that a peak is skipped if the part of the history immediately following it has already been paired with a previously counted range. If the initial peak of a range is a minimum, a cycle is counted between this minimum and the most positive maximum which occurs before the strain becomes more negative than the initial peak of the range. For example, in Figure 1 a cycle is counted between peak 1 and peak 8, peak 8 being the most positive maximum that occurs before the strain becomes more negative than peak 1. If the initial peak of a range is a maximum, a cycle is counted between this maximum and the most negative minimum which occurs before the strain becomes more positive than the initial peak of the range. For example, in Figure 1 a cycle is counted between peak 2 and peak 3, peak 3 being the most negative minimum before the strain becomes more

BEST AVAILABLE COPY

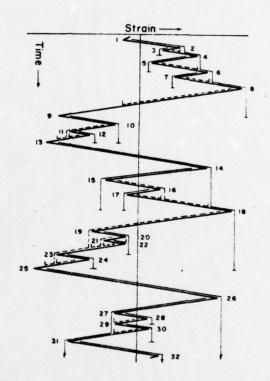


Figure 1

Illustration of Range-Pair Counting Method

positive than peak 2. Each range that is counted is paired with the next straining of equal magnitude in the opposite direction, explaining why complete cycle rather than half cycle counts are made. For example, in Figure 1 part of the range between peaks 8 and 9 is paired with the range counted between peaks 1 and 8.

C. DAMAGE CALCULATION

Damage due to plastic strain and damage due to elastic strain is calculated separately. The behavior of the specimen is assumed to be elastic-perfectly plastic as illustrated in Figure 2. Behavior in compression is assumed to be the same as the behavior in tension. The total damage is determined using the Miner's summation $D = \sum n/N$ where

- D = total damage
- n = number of cycles at a particular load level
- N = number of cycles to failure at a particular load level

Failure of the specimen is defined as occuring when D equals unity.

Plastic strain damage is calculated using the Coffin-Manson theory [Ref. 5]. The Coffin-Manson theory is based on an experimentally determined log-log plot of plastic strain range vs. cycles to failure as shown in Figure 3. The slope of the line in Figure 3 is approximately -1/2 for all metals, with the ordinate intercept making the behavior of each metal unique. The equation used for calculation of plastic strain damage is

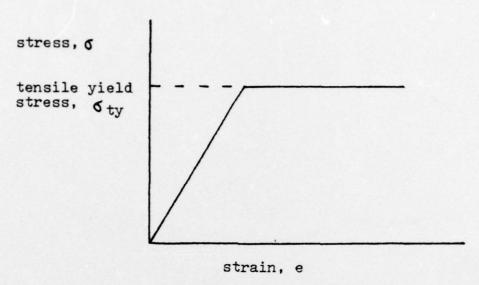


Figure 2. ELASTIC-PERFECTLY PLASTIC BEHAVIOR

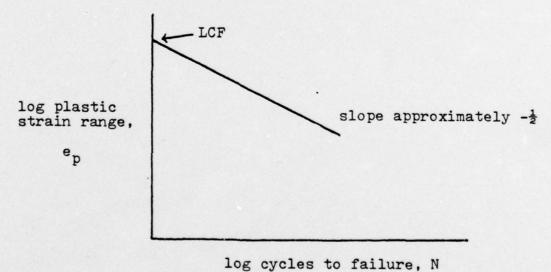


Figure 3. COFFIN-MANSON DIAGRAM

 $N = (e_p / c)^{k-1}$

where N = number of cycles to failure at a specific plastic strain

 e_{p} = magnitude of the plastic strain

k = slope of the Coffin-Manson curve, approximately -1/2

c = ordinate intercept; experimentally for each metal
 type

For strain levels in the plastic range, failure occurs in less than 10⁴ cycles, and damage is calculated using the Coffin-Manson theory. For strain levels in the elastic range, the Goodman diagram is the basis for calculation of damage. The maximum and minimum local strain levels are sequentially compared with second order least-squares curve fit data from the Goodman diagram. The curve fits used are for lives of 10⁴, 10⁵, 10⁶, and 10⁷ cycles [Ref. 3]. The elastic damage is totaled using the Miner summation and combined with the plastic damage, computed in a similar way, to yield the total damage. The amount of plastic damage and the total damage for a particular flight, and the total damage including all previous flights is printed at the end of each flight.

III. DISCUSSION OF RESULTS

The load spectrum used in this analysis is the load spectrum A from MIL-HDBK-5A, and the loads are assumed to be applied at wing station 32 on an A7 aircraft wing [Ref. 2]. Load data were input as block loads and also as single randomized cycles. The Miner's Law summation of fatigue damage for this loading spectrum is .1300 per 1000 flight hours. Results obtained using this computer program are given in Tables II, III, and IV. It is important to notice that when fatigue damage is calculated without using the range-pair counting method, it makes only a minor difference on the total fatigue damage when the sequence of loading is changed or when negative loads are included.

Block loads were arranged in LO-HI, HI-LO, and HI-LO-HI sequences. The results indicate the LO-HI sequence to be more damaging than the HI-LO sequence, with damage due to the HI-LO-HI sequence falling between the other two block load sequences. In the HI-LO sequence the HI loads leave a local residual compressive stress in the material, and the LO loads do less damage.

In the case of RANDU generated load sequences, identical starter integers were used in order to evaluate the effect on the damage of inclusion of MIL spectrum A negative loads and ground cycles. The more negative loads that are included, the worse is the fatigue damage. The increase in damage with more negative loads is caused by the negative loads decreasing

TABLE II

FATIGUE DAMAGE DUE TO POSITIVE MILSPEC A BLOCK LOADING, NORMALIZED TO 1000 HOURS

SEQUENCE OF BLOCK LOADS	USING RPCM*	NOT USING RPCM*
LO-HI	.16780102	.16780108
HI-LO	.069243014	.069243014
HI-LO-HI	.10864216	.10839599

^{*} RANGE-PAIR COUNTING METHOD

TABLE III

FATIGUE DAMAGE DUE TO MILSPEC A RANDOMIZED LOAD INPUTS, USING RPCM, NORMALIZED TO 1000 HOURS

TO START RANDU	MINIMUM LOAD VALUES			
	11 PERCENT LIMIT LOAD	NEGATIVE MILSPEC A	GROUND CYCLES	
83745,54711,54487	.039226338	.046267733	.060848035	
13547,66549,7	.095124505	.085063167	.10429338	
9,583,4777	.043321513	.043640956	.036781987	
48621,3,491	.018672124	.034077277	.035881512	
73,559,1001	.031819174	.042006299	.037130679	
357,833,1	.037665060	.049336702	.050973855	

TABLE IV

FATIGUE DAMAGE DUE TO MILSPEC A RANDOMIZED LOAD INPUTS, NOT USING RPCM, NORMALIZED TO 1000 HOURS

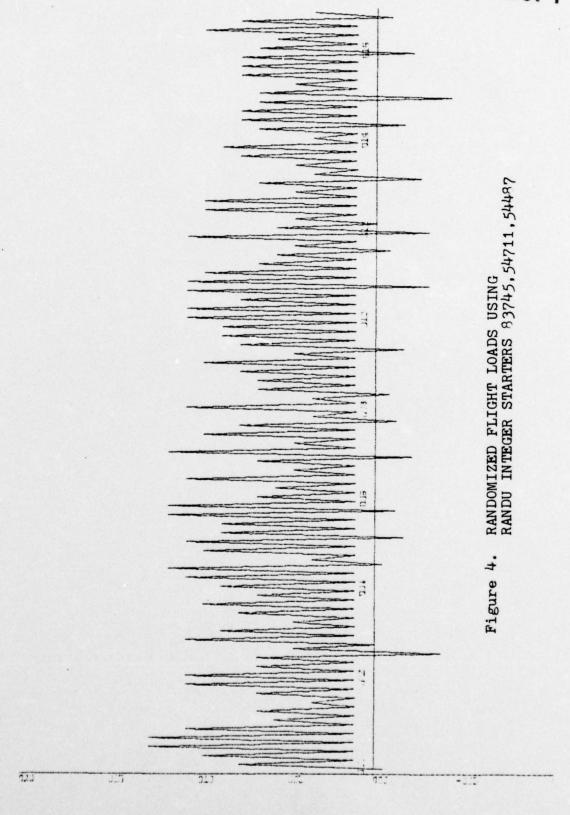
INTEGERS USED TO START RANDU	MINIM		
	11 PERCENT LIMIT LOAD	NEGATIVE MILSPEC A	GROUND CYCLES
83745,54711,54487	.040602274	.047398917	.049124956
13547,66549,7	.040137805	.047727041	.049612150
9,583,4777	.041204430	.048346408	.049540661
48621,3,491	.041160211	.049230121	.050957575
73,559,1001	.040792227	.047043338	.049311966
357,833,1	.041056350	.049220584	.050481893

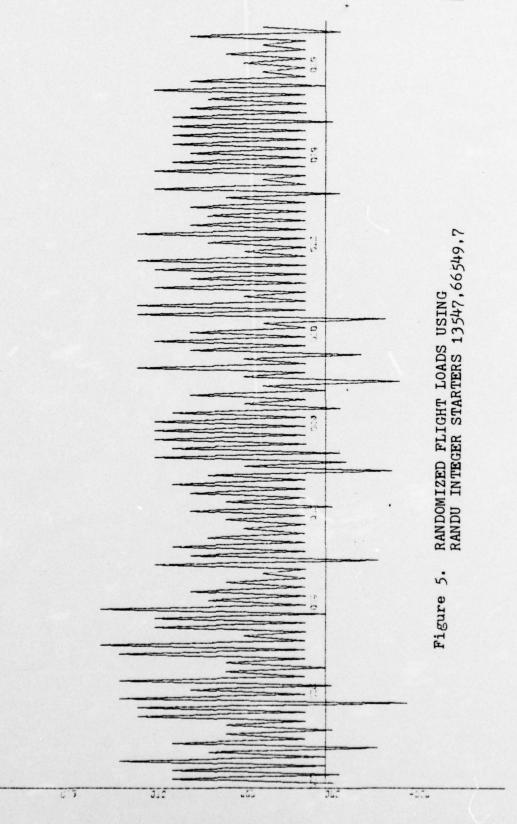
the local residual compressive stress caused by positive loads into the plastic zone. This permits the local positive elastic stresses to be more damaging.

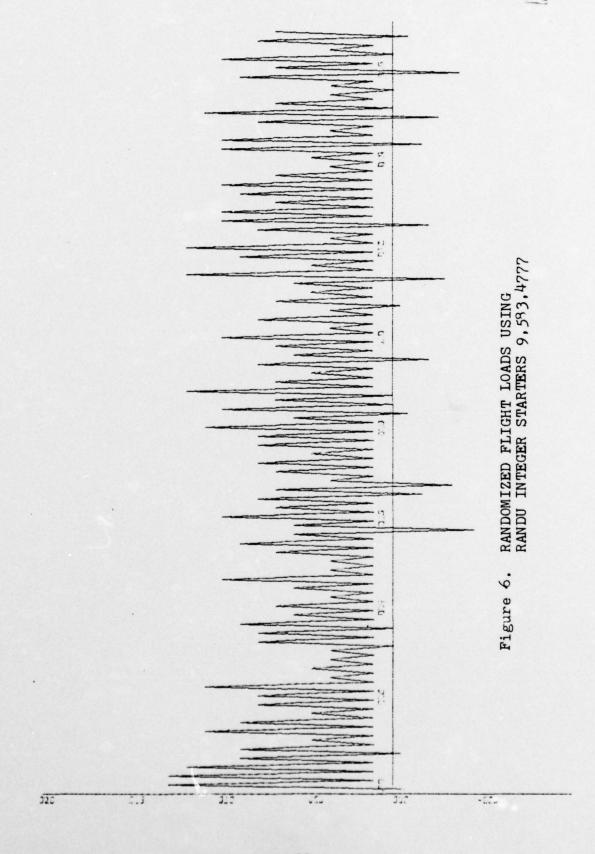
Six plots are presented, beginning on page 33, to illustrate the pattern of the individual load randomization process. Each plot portrays the first two flight hours of the aircraft. The range of values in the columns of Table III are caused by such variations in the load randomization pattern.

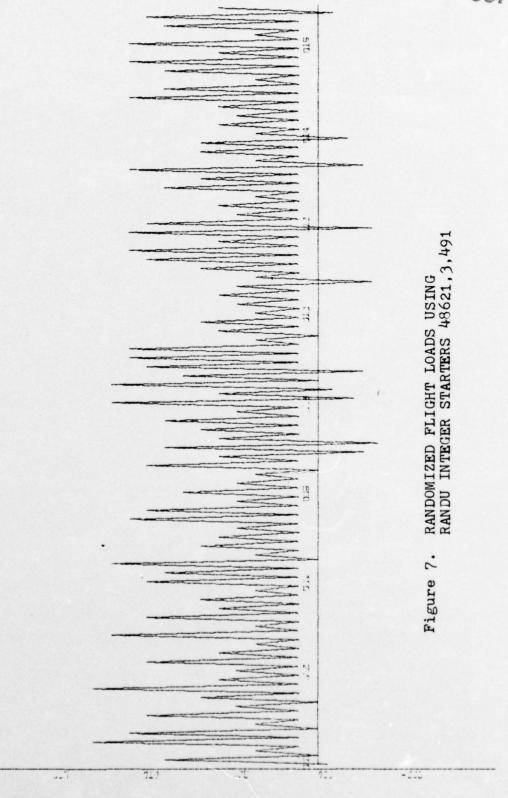
The fatigue damage resulting from randomization of loads on a cycle basis, as shown in Table III, is significantly different from the damage due to the HI-LO-HI block loads illustrated in Table II. Vought Aeronautics Division conducted a fatigue study using randomized flights. Each flight was based on a mission profile specified by the Air Force. Fatigue damage resulting from the randomized flights was in close agreement with fatigue damage resulting from HI-LO-HI block loading.

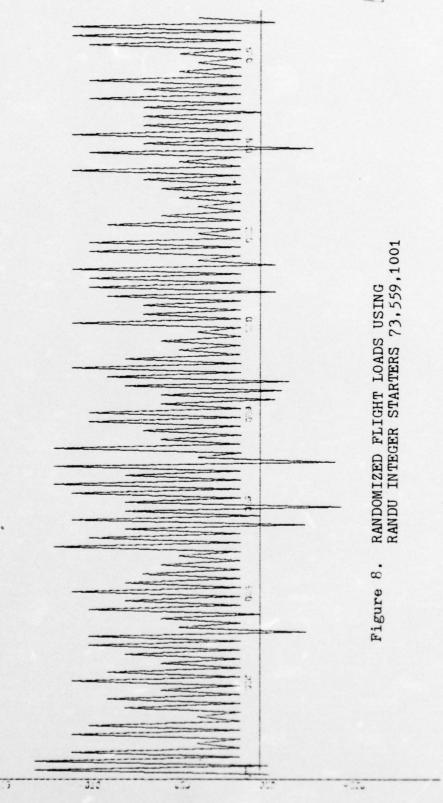
BEST AVAILABLE COPY

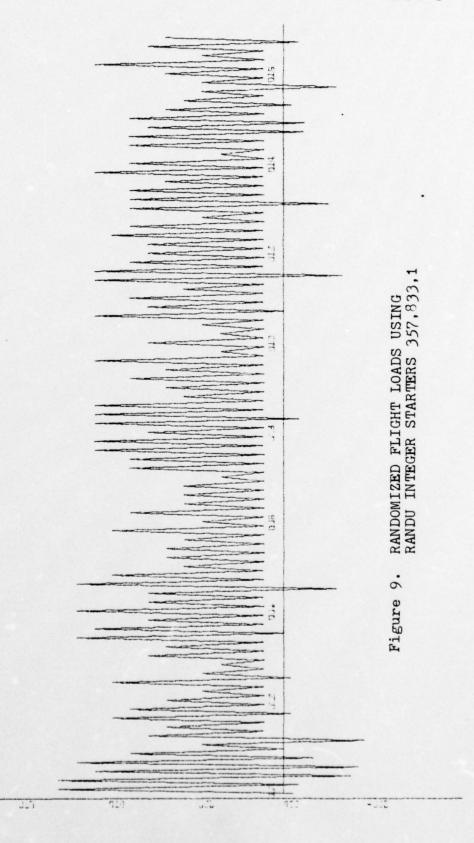












32.1

IV. RECOMMENDATIONS FOR FUTURE RESEARCH

Subsequent work in this area should be directed toward refining the input load sequence. The Naval Air Development Center collects accelerometer data from individual aircraft on a monthly basis. These load data could be used in this computer program to see if randomizing the load sequence over the life of the structure provides a different picture of the fatigue damage than randomizing by the month. Also, research is needed to determine the magnitude and sequence of the ground cycles. In this analysis all ground cycles were considered of equal magnitude and equal spacing, which is not the case with a real airplane. A landing load spectrum should be used to be more accurate.

The computer program itself could be improved by a better understanding and definition of the relaxation phenomena. Experiments to quantify the relaxation parameters will enhance the program.

The Air Force is monitoring crack growth as the basis for their fatigue damage assessment. It would be productive to compare the total fatigue damage calculated by the two methods.

```
1075
7075
7075
7075
          4000
        C.4
71.7MIL-HDBK-5A
56.3MIL-HDBK-5A
44.6MIL-HDBK-5A
38.1MIL-HDBK-5A
1.
                       101
                                        25
                       83
41
                       10
                                        75
         -00020
-00014
-00114
-0013
100000000
2.72
2 29.8
                                        ma
                                        50
                    A7 CRITIC
                        1/10/10/00/00
```

OUTPLT CF BLOCK LOADS, IPRINT = 1

SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

					DATA	71.70000 56.29999 44.55599 38.09999			E2 = 1.000			150.00000000000000000000000000000000000
FROM U166808	72.00000	000	-1.83600	*	JARE FIT OF S-N	0.51540 0.51540 0.61410 0.68380			1.000 AND	10 LOADS		X X X X X X X X X X X X X X X X X X X
A7 CRITICAL POINT #1 - 7075-T6 AL	TRESS (KSI)	INTERCEPT = 0.40000	COFFIN-MANSON SLOPE	= 10000.000000	SECOND ORDER LEAST SMIN*2 + B(I)*SMIN	-0.00200 -0.00220 -0.00140 -0.00130	PLIED FROM MIL-HOBK-5A	RELAXATION FUNCTION	C1/(KTSMAX**E1 * KTSMEAN**E2) = C.10000000E 08 , E1 =	THROUGH BLOCK OF	= 29.79999	XT
SPECTRUP FRCM MATERIAL TYPE -	TENSILE YIELD STRESS	LCF STRAIN INTE	INVERSE CF COFF	ELASTIC MCDULUS	COEFFICIENTS OF	1000 1000 1000 1000 1000 1000 1000 100	UNNOTCHEC COUPON S-N INFCRMATICN SUPPLIED	RESIDUAL STRESS	WHERE CI = C.	2 TIMES	LOAD LIMIT	STEP 14 PE 14 PE 14 PE 15 PE 16 PE 1

TYPE

BLOCK TYPE TYPE TYPE TYPE TYPE TYPE TYPE

ma

0-

SPECTRUM FRCM AT CRITICAL POINT #1 FRCM U166808

2.72 AKT =

		40		12	22	32
		77		m	4	ď
		400		40	40	64
	Ē	0.17906543E 0.21008384E		C.24992734E	0.30229507E	0.37305303E
	TECHNIQUE	16.00 40.00	444444444	רשושושושושושוש	RIRIO44444	44444444444444444444444444444444444444
	CCUNTING	EQRES -29.32		-13.11	-5-00	0.0
10000000000	RANGE-PAIR	2000	1111111 2000000000 2000000000 2000000000	-19.95 -19.95 -19.93 -19.10 -18.90 -18.90	-18.30 -18.10 -17.65 -16.25 -15.58	-14.32 -13.72 -12.58 -12.03 -10.93 -9.33
CONSTANT C1=	2	72.00 63.89	00000000000000000000000000000000000000	00000000000000000000000000000000000000	00000000000000000000000000000000000000	00000000000000000000000000000000000000
RELAXATICA CON	- 0	4	ANNER COLOCY AND		00 00000	RELAXATION RELAXATION RELAXATION RELAXATION 25.33 3.28 RELAXATION RELAXATION

00000000000000000000000000000000000000		
4 7400 0 100 100 100 100 100 100 100 100 100	COUNTE	000000000000000000000000000000000000000
136.00 136.00 136.00 136.00 136.00 136.00 136.00 2550.00 2	INPUT LOAD SPECTRUM = 55 SIGMA MINIMUM	
888866557444876768888866671748888888888888888888	EYS IN THE AXIMUM	0.6440000000000000000000000000000000000
RECLAXATION RECLAXION RECLAXATION RECLAXATION RECLAXATION RECLAXATION RECLAXAT	E NUMEER OF PEAKS O	1111111 8765549010984655490

44444444444444444444444444444444444444	00* 00* 00* 00*	COUNTER K	16.00000
10000000000000000000000000000000000000	863383E C 890269E 0	SIGMA	-0.2C4038E 02 -0.2C33860E 02 -0.2C3507E 02 -0.2C3154E C2
00000000000000000000000000000000000000	363929E 0 283561E 0 RANGE PA	MAX IMUM	0.720000E 02 0.639121E 02 0.639475E 02 0.639827E 02
1000000000000000000000000000000000000	14v	STEP	-0 64

できてのちょうらからからできていちもくていちもしゅうからていちもしゅうらからからしていちょうなんかかかかからををををををををしてころろろろろろことにしてしてしてしてして

6500° 00000 9500° 00000 8****	LIFE CAMAGE 0.603661E-03 INS= 0.60366094E-03	日本の
	J A	000000000000000000000000000000000000000
000		
	IGUE MIN STR	<u> 4 ԹԱՐԱՐԻՐԻՐԻ ՄԱՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻՐԻ</u>
600	IGU MIN ST	000000000000000000000000000000000000000
000	- 0	<u> </u>
5000	A XXH	00-00000000000000000000000000000000000
583	T OAT	くらしてしているというというというというというというというというというというというというというと
ထာထာထာ	DXQ	ほりしましなしなっちょうであるうろのらんりらてしてのろてきるのよしき
	TIN MA.	しゅうしゅうしゅうしゅうしょうしょうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅうしゅう
000	E E G	~ 000000000000000000000000000000000000
	J E	しりしょしししししょしゅうこうししししし サイヤイレー こうしょうしょうしゅうしゅ
	SUL	
	ESUL	000000000000000000000000000000000000000
	Œ	
16 02 16 02 16 02	STRAINS WITH IC STRAIN 0.00706 DAMAGE	Unining
000	æ 0	
330	100	
4000	-	
440	PLASTIC PLAS	
000	PLA	
	SA	
	2	NOONUBATHTBUTTOOOOOOOUTUBTUATHTBUTTOTT
	۵.	14 14 15 15 15 15 15 15
	0	00000000000000000000000000000000000000
	AND	
	4	ω,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
	S	Wa
440	S d-	NO NO NO NO NO NO NO NO
S	ES	Σ
	ST	コーセともももももももられるちららららららららららららららららららららららららららららら
	STRES	()
	U)	
	_	
	LOCAL	
	20	
	-	

	-72	m
;	24	70
0 2	0.21008384E	0.24992734E
rc = 0.	14 004444444444	
L ENN/C	-21-21	-13.11
TOTA	-29.32	-28.97
2 .51	00000000000000000000000000000000000000	88888888888888888888888888888888888888
NO. SIGMAX	00000000000000000000000000000000000000	00000000000000000000000000000000000000
FLIGHT CR BLOCK	NN	THITITITIES OF THE PARTY OF THE

0.22347500 0.22347500 0.223540562750 0.22354056750 0.12361424035611002 0.1236141233611002 0.1236141233611002 0.2222354611002 0.2212216466111002 0.22225546111002 0.22225546111002 0.2222566111002 0.2222566111002 0.2222566111002 0.2222566111002 0.2222566111002 0.2222566111002 0.2222566111002 0.22225666111002 0.22225666111002

000000000000000000

шишшшшшшшшшшшшшшшшшш

4 301

9

0 11 SET

S I

ER 0

MAGE

*NUNNUNUNUM MWWWWWW*44 444444400 NW4N0LB001 NW4N0LB001 NW4N0LB001

0.302255C7E C4 4 22	0.37305303E C4 5 32	0.47195547E C4 6 42	0.61621797E 04 7 52 0.83860000E C4 8 53 0.12079809E 05 9 54 0.18907535E 05 10 55	ATIGUE LIFE	MIN 0.603239E-03 STRAINS= 0.60323928E-03	04 C.16000031E-02 05 C.36393036E-03
00 144444444444444444444444444444444444	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	* 000000000000000	**** ****	SULTING F	MAX OR NAX ROM PLASTIC	0.99559E05E 0.10591109E
OSONDOMPANO	20 20 20 20 20 20 20 20 20 20 20 20 20 2	1004400Lumun-	1048C	STRAINS WITH RE	IC STRAIN 0.00706 DAMAGE F	RNCYC 16.
000 1	1100 L 0 C 4 C C C	7-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4	-049 -049 -049 -049 -049 -049 -049 -049	AND PLASTIC S	PLAST	516MIN -20.40 -20.39
		TOUCH TOUCH TO THE TOUCH TO THE TOUCH TO THE TOUCH TO THE TOUCH TO	19-37 16-37 18-39 10-43	LOCAL STRESSES	STEP 1	SIGMAX 72.00 63.91

 $\frac{1}{2} \frac{1}{2} \frac{1}$

0.11300¢65E 06 0.55206530E-02 0.77923370E 07 0.83415280E-03 0.20005€53E 09 0.47486101E-04 0.47726305E 10 0.35619769E-05 DAMAGE PER THIS SET= 0.69242597E-01 TOTAL ENN/CYC =, 0.13848418E 00

4500. 6500. 9500. 17000.

9888 98438 98438 99338

57.09 24.005 28.39 28.39

```
1075
7075
7075
7075
             4000
          C.4
71.7MIL-HCBK-5A
56.3MIL-FCBK-5A
44.6MIL-HDBK-5A
38.1MIL-HCBK-5A
                              10
                              3
1 FROM U166808
72.
2801
5154
6141
6838
                                                    41
                                                    90
                         10
                             74
          -.0020
-.0022
-.0014
-.0013
100000000
2.72
2 29.8
     A7 CRITICAL POINT #1
                                                   ma
                         14 -14W4RIAL@ROOT
                              してきょうしゅいい
```

OUTPLT OF BLOCK LOADS, IPRINT = 1

1 CATA DECKS ARE TO BE FROCESSED.
SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

```
1.000
                                                         71.70000
56.29599
44.555999
38.09999
                                                DATA
                                                                                                                =
                                                                                                              E2
                                                 N-S
                                                                                                                                                                                                           TYFE
                                                                                                              1.030 ANC
                                                                                                                                                OF
#1 FRGM U1668C8
                                                                                                                           LOADS
                              -1.83600
             72.00000
                                                 FIT
                                                SQUARE FIT

(1)

0.28010

0.28010

0.51540

0.63380
                                                                                                                            10
                                                                                                                                                                                                           TYP
                      0.40000
                                                                                                         KTSMEAN**E2)
                                                                                   PATA DERIVED FROM FROM MIL-HOBK-5A
                                                = A(I)*SMIN**2 + B(I)*SMIN +
-0.00200
-0.00220
-0.00140
-0.00130
                                                                                                                                                                                                           TYPE
                                        10000.00001
                                                                                                 RESIDUAL STRESS RELAXATION FUNCTION
                                                                                                                                   29.79999
-7075-T6 AL POINT
                                                                                                                           THROUGH BLOCK OF
                                                                                                                                                                                                           TYPE
                                                                                                                                                 SLOPE
                                                                                                         *0
                                                                                                                                                                                                          TYPE
             (KSI)
                                                                                                         = 0.10030000E
                               CF COFFIN-MANSON
                                                                                                                                                                                                          TYPE
                                                                                   UNNCTCH EL COUPON S-N
INFCRMATICN SUPPLIED
             TENSILE YIELD STRESS
                     LCF STRAIN INTERCEPT
                                                                                                                                                                                                          TYPE
                                                                                                                           TIMES
                                        MODOLUS
SPECTRUN FRCM
                                               COEFFICIENTS
SMAX = A
LIFE
10** 5
10** 7
                                                                                                                                                                                                          TYPE
                                                                                                                                                "5
                                                                                                                                    LOAC
                                                                                                         WHERE (
                               INVERSE
                                       ELASTIC
                                                                                                                                                                                                          BLOCK
                                                                                                                                                 STE
```

-0

20

mæ

41-

92

20

-+

ma

50

2

SPECTRUM FRCM AT CRITICAL PCINT #1 FRCM U166808

AKT =

0m4500r			0490 100870
-72	15	22	32
72	m	4	S
44	4	4	4
0.17906543E	C.24992734E	0.36229507E	0.37305303E
200000000	44440m/m/m/m/m/m	44444444444444444444444444444444444444	44444444
EQRES -29.32	-13.11	-5.30	0.0
00000000.00 6MIN RES 40 -29.32 335 335 228 228	-28.97	-26.92	-20.68
1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	1110000	11111111111111111111111111111111111111	4mm201021-0
→ → → → → → → → → → → → → → → → → → →	144400000		W44W00000H4
BBSC SSHOON SSHO	.28	. 28	. 28
RELAXATION RELAXATION RELAXATION RELAXATION RELAXATION RELAXATION	TO T	TOUCO COCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOCOC	XXXX@XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX

PC000-10	こうようらて 色り 〇1
wwww44	444444400

0.47195547E 04 6 42	0.61621757E 04 7 52 0.83860000E 04 8 53 0.12079809E 05 9 54 0.18907535E 05 10 55	TIGUE LIFE	MIN 0.603661E-03 STRAINS= 0.60366054E-03	055 055 055 055 055 055 055 055 055 055
0 0 0 0 0 0 0 0 0 0 0 0 0 0	*****	RESULTING FATI	MAX CR MAX FROM PLASTIC	00000000000000000000000000000000000000
111123478 201122 20112	22-47480	TIC STRAINS WITH	PLASTIC STRAIN 0.00706 DAMAGE	RNC 1007 1007 1007 1007 1007 1007 1007 100
20000000000000000000000000000000000000	70000mm	AND PLASTIC	٩	8 111111111111111111111111111111111111
RRELLAXXA RECLAXXA RE	16 - 39 TIC	LOCAL STRESSES	STEP	000000011404400000 XO40000011404400000

```
2210222224619

2442224619

251233466619

251233466619

251233466619

25123346619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

2512334619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

251234619

                                                                                                                                                                                                                            12
                                                                                                                                                                                                                      48E
                                                                                                                                                                                                   4E
                                                                                                                                                                                                  301
                                                                                                                                                                                                                       348
384N
                                                                                                                                                                                                                           906
                                                                                                                                                                                                  $24
924301
                                                                                                                                                                                                                           0.179
 9
                                                                                                                                                                                                   0
9
                                                                                                                                                                                                   =
                                                                                                                                                                                                                      E
                                                                                                                                                                                                             0
                                                                                                                                                                                                  S
                                                                                                                                                                                                              11
                                                                                                                                                                                                  S
                                                                                                                                                                                                                            7
                                                                                                                                                                                                  I
                                                                                                                                                                                                            ENN/CYC.
                                                                                                                                                                                                                       SN-
                                                                                                                                                                                                                      WMN
                                                                                                                                                                                                  ER
                                                                                                                                                                                                                      E08
                                                                                                                                                                                                            OTAL
                                                                                                                                                                                                  MAGE
                                                                                                                                                                                                                      510
                                                                                                                                                                                                                      MOM
                                                                                                                                                                                                                      29.
                                                                                                                                                                                                   DA
1
                                                                                                                                                                                                                       Z
                                                                                                                                                                                                                      200000

244 www

200000
                                                                                                                                                                          4000
                                                                                                                                                                                                                            NUNNN
                                                                                                                                                                                                                            11111
                                                                                                                                                                                                                       400 HUS
                                                                                                                                                                                                                 NO. SIGMA
72.006
63.91
63.95
63.95
×zww
33.22
3.22
3.22
                                                                                                                                                                                                                      S
                                                                                                                                                                                                                                     777
                                                                                                                                                                                                                                     200
                                                                                                                                                                                                                  J
                                                                                                                                                                                                                FLIGHT
31MAX
34.25
RELAXAT
RELAXAT
```

Vm4

000 000 000 000 000 000 000 000 000 00
000 000 000 000 000 000 000 000 000 00
000 000 000 000 000 000 000 000 000 00
000 000 000 000 000 000 000 000 000 00
000 000 000 000 000 000 000 000 000 00
000000000000000000000000000000000000000
-
444444000000000000000000000000000000000
1 4 キューニューニー・ 4 カーニーニュー・ 2000/00/00/00/00/00/00/00/00/00/00/00/00
00 0 0
93 68 95
8 9 0 8
777777777777777777777777777777777777777
44444449999000000000000000000000000000
.28 .28
ZZZZZZ ^m ZZZZZZZ ^m ZZZZZZZZ ^m ZZZZZZZZ ^m ZZZZZZZ ^m ZZZZZZZ
A PAPARA PARA PARA PARA PARA PARA PARA

6	0	-
3	2	2

0.61621797E 04 7 52 0.83860000E 04 8 53 0.12075809E 05 9 54 0.18907535E 05 10 55	IGUE LIFE MIN 0.603239E-03 STRAINS= 0.60323928E-03	
**************************************	SULTING FAT MAX CR ROM PLASTIC	できまれるようなできるとうなっていまます。 りのものでもなってもなるのできるともできることできるのできるのできるのでもあった。 ならてもなのでららもこくののしましょうのかなるをはなります。 も目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目目
233 243 264 264 264 264 264 264 264 264 264 264	STRAINS WITH RE	NI Oo4444444444440000000000004444444444400000
286.3999888	AND PLASTIC	00000000000000000000000000000000000000
RELAXATION RELAXATION 19.37 16.29 16.29 13.41 13.41 13.43 10.43	LOCAL STRESSES STEP	0.000000000000000000000000000000000000

16 652425 3848418E 0 11 SET աաաաաաաաաաաաաաաաաաաաա 0 S THI ER MAGE TOT

らろうようなできらいまままらららららららららままする ころうならないといるといるようなしてしているとしているとしている。 もしなてものできることできることできることできることできることである。 らることできることできることできることできる。

```
100000.
7075
7075
7075
                  4000
              71.7MIL-HDBK-5A
56.3MIL-HDBK-5A
44.6MIL-HDBK-5A
38.1MIL-HDBK-5A
                                     145000
245000
250000
250000
190000
                                                             10
                                    ma
47 CRITICAL PCINT #1 FROM U166808
7075-16 AL -.0020
-.0022
-.0022
-.0022
-.0022
-.0022
                                                             41
                                                            92
                               10
                                    10
                                                            38
                29.8
                                                            50
                               0 1004504F85001
                                    しいちょうしまらり
```

OUTPLT CF BLOCK LOADS, IPRINT = 2

1 CATA DECKS ARE TO BE PROCESSED.
SPECTRUM SUBJECTED TO THE RANGE-PAIR CCUNTING TECHNIQUE SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

```
71.76000
56.29999
44.559999
38.05999
                                                   DATA
                                                                                                                       11
                                                                                                                      E2
                                                   NIS
                                                                                                                      1.000 ANE
                                                                                                                                                          OF
POINT #1 FRCM U1668C8
                                -1.83600
             72.00000
                                                   FIT
                                                  SECOND ORDER LEAST SQUARE FITSMIN**2 + B(I)*SMIN + C(I)

A(I)

-0.00200

-0.00220

-0.00140

0.51540

0.61410

0.68380
                                                                                                                                   10
                       0.40000
                                                                                                               KTSMEAN**E2)
                                                                                         UNNCTCFEC COUPON S-N DATA DERIVED FROM INFCRMATICN SUPPLIED FROM MIL-HOBK-5A
                                                                                                                                             29.79999
                                          10000.00001
                                                                                                       STRESS RELAXATION FUNCTION
                                                                                                                                   BLOCK OF
                                                                                                                                                               0000000000
                                SLOPE
                                                                                                                 *0
                                                                                                                                                           -A CRITICAL
             STRESS (KSI)
                                                                                                                = 0.10000000E
                                COFFIN-MANSON
                                                                                                                                   THROUGH
                      LCF STRAIN INTERCEPT
                                                  COEFFICIENTS OF SE
SMAX = A(I) * SN
LIFE
10 * * 5
10 * * 5
10 * * 7
                                         P G D U L U S
                                                                                                                                   2 TIMES
             YIELD
SPECTRUM FRCM
                                C.F.
                                                                                                                 "5
                                                                                                                                                           日上ろうようらりのりつ
                                                                                                                                            LCAC
                                                                                                                                                           TYPE
                                                                                                             WHERE
                                                                                                       ESICUAL
             TENSILE
                                INVERSE
                                         ELASTIC
                                                                                                                                                           日上ころはちらりの
                                                                                                                                                          STE
```

TYPE

TYPE

TYPE

TYPE

TYPE

TYPE

TYPE

TYPE

BLOCK

ma

1-4

æm

SIV

			JE .	NEP 0.17906543E 0.21008384E 04	C.24992734E 04	0.302295C7E C4	0.37305303E 04
CM U166808			CCUNTING TECHNIQUE	200000000	144440 <i>ajajajajaj</i>	00 00 00 00 00 00 00 00 00 00 00 00 00	44444
CRITICAL POINT #1 FRC		T C1= 10000000.00	D THE RANGE-PAIR	MAX 8991 12003 9951 12003 9951 12003	20000000000000000000000000000000000000	7.50 7.50 7.50 7.50 1.80	65555555555555555555555555555555555555
SPECTRUM FRCM AT	AKT = 2.72	RELAXATICA CONSTANT	ECTRUM SUBJECTED	STMAX STMIN STMAX STMAX STMAX STMIN STMAX STMIN	THE LAYS AND THE L	NANSANANANANANA	ELAXATICA ELAXATICA ELAXATICA ELAXATICA ELAXATICA ELAXATICA ELAXATICA ELAXATICA FILAXATICA

m

400mmm44 444444400 400mm01 000mm001

.47195547E 04 6 42	•61621797E 04 7 52 •838600C0E C4 8 53 •12079809E 05 9 54 •18907535E C5 10 55	DAMAGI 0.603661E-03	CC. 200 CC. 20
	0000	ATIGUE R MIN X X STR	
######################################	0000****	TING F	10000000000000000000000000000000000000
0.0	0000	RESUL	000000000000000000000000000000000000000
-8.93	100.00 100.00 100.00 100.00	STRAIN 0706	
	2000 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	PLASTIC	
00000000000000000000000000000000000000	00000m	AND PLA	10000000000000000000000000000000000000
SNNNNN W		STRESSES STEP	00000000111000400000000000000000000000
	0969CLCC 0969CLCCC	LOCAL	

72 α HO C 00 **ப**யய 4 MW4 3014 Z400 Sm 98 524 4301 00 .21 9 55 00 0 9 ET ,un 11 44 YC C SNI ENN 9.3 1.2 ш ヨスマ MAGE AL OT/ 570 MOM -0-29. 1 Z SIGMI 0.40 0.39 4001 NNN 1 1 1 400. 16M No. 200 UZOO OHNN BA TE S -XUL4 XVVVIDI DE . . A

	17	22	32	45			
	m	4	w	•			
	40	40	42	64			
	3 t	.7E	E .	7E			
	273	956	530	554			
	.249927	022	730	.4719			
	6.2	m.	m	4.0			
		0	0				
00000000		0044444444	00000000000000000000000000000000000000	040000 * • • • • • • • • • • • • • • • • • • •			
	=	00	0	0			
	<u>m</u>	r.	0	0			
	ī						
	16	92	8 9	63			
	28.	56.	50.	.8			
	T	1	ï				
000001110	1440804777 1474	300110mm440 30418011-wg/	04400000000000000000000000000000000000	1200445			
	.28	28	28	28			
ZZZZZZZZ		e ZZZZZZZZZ					
XXXXXXXXX	XXXXXXXXXXX	CONTRACTOR DE	PPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPPP	XXXXXMX			
աատատաատա	յա տուտոտատարո տ	ոտատաաատաատու		12mmmmin			

からからろ しりめとからかをで このらめとうられをこ このらもしりらって サイヤイ サイをををををををを

0.61621797E 04 7 52 0.838600C0E 04 8 53 0.12079805E 05 9 54 0.18907535E 05 10 55	IGUE LIFE MIN 0.603239E-03 STRAINS= 0.60323928E-03	00552222222222222222222222222222222222
****00000 ***** *********************	MAX CR PLASTIC	4648624468888999999999999999999999999999
0000	RESUL E FROM	000000000000000000000000000000000000000
-22 -0.49 -0.08 -0.08	STRAIN 0706 DAMAG	Δα Ο 0 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4
44/0/0/0/0/0/0/0/0/0/0/0/0/0/0/0/0/0/0/	PLASTIC STR	
2867-272 864-0098 2864-0098 2864-0098	AND PLA	11111111111111111111111111111111111111
RELAXATION RELAXATION RELAXATION 19.37 ICN 19.37 ICN 13.439 103.431 103.431 103.431 103.431 103.431 103.28	LOCAL STRESSES STEP	0.000000000000000000000000000000000000

56 52425 .1384641 0 0 E1 . = DAMAGE TOTAL

```
100000.
7075
7075
7075
              4000
          C.4
71.7MIL-HCBK-5A
56.3MIL-HCBK-5A
44.6MIL-HCBK-5A
38.1MIL-HCBK-5A
11.
                               10
                               83
*I FROM U166808

*2801

*2801

*5154

*6141

*6838
                                                      62
                               1-5
           -0020
-0022
-0014
-0013
100000000
      A7 CRIT CAL PCINT #1
                                                     38
                                                     50
                          0 10014010L@0001
                               1234500000
```

OUTPLT CF BLOCK LOADS, IPRINT = 2

NO CCUNTING METHCDS USED SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

```
1.000
                                                                 71.70000
56.29599
44.59599
38.05999
                                                       S-N DATA
                                                                                                                               11
                                                                                                                               E2
                                                                                                                              1.000 ANC
                                                                                                                                                                      CF
SPECTRUM FRCM A7 CRITICAL POINT #1 FRCM U166808
MATERIAL TYPE -- 7075-T6 AL
                                  -1.83600
                                                                                                                                            10 LOADS
              72.00000
                                                       FIT
                                                     TYPE TYPE TYPE TYPE
                        0.40000
                                                                                                                        KTSMEAN**E2)
                                                                                               UNNGICHEC COUPON S-N DATA DERIVED FROM INFORMATION SUPPLIED FROM MIL-HOBK-5A
                                                                                                              RESIGNAL STRESS RELAXATION FUNCTION
                                                                                                                                                       29.79999
                                             10000.00000
                                                                                                                                            2 TIMES THROUGH BLOCK OF
                                                                                                                                                                           0000000000
                                   SLOPE
                                                                                                                        ENEP = C1/(KTSMAX**E1 * WHERE C1 = C.1000000E 08
                                                                                                                                                                      ×0000000000
              TENSILE YIELD STRESS (KSI)
                                   INVERSE CF COFFIN-MANSON
                                                                                                                                                                                                                                        TYPE TYPE TYPE
                         LCF STRAIN INTERCEPT
                                            ELASTIC MODULUS
                                                                                                                                                       LIMIT
                                                                                                                                                                      TYPE
                                                                                                                                                                               LCAE
                                                                                                                                                                                こうちらり もらり
                                                                                                                                                                      STE
```

TYPE

BLOCK

101

20

mæ

31

00

20

1-3

ma

50

9-

-2

SPECTRUM FRCM AT CRITICAL POINT #1 FRCM U1668C8

AKT =

*			
-2	12	22	35
-72	m	4	'n
44	4	4	4
00	Ö	ò	ò
mw4 7 mm	4	C7E	35
748 248	.2499273	n	30
000	266	529	305
210	546	305	373
00		•	•
2000000000	200000000	000000000000	000000000
ma0444444	2440ninininin	Uninininini024444	44444
-44	-	4	*
23.2 1.25.8	=	00	•
86-		· .	•
ш77 1	1	1	
0000000.00 6MIN RES 40 -29.32 332 232 218	9.1	92	8 9
0 406	28.	56.	20.6
N N N	ï	ï	ï
O YOOGUNBUID	4-r-95000-c	000000-W4W@4	-0000000000000
	11000rum-	1004W1000VNQ	シアーちのてのきめら
NUNNUNUNU	ンろろろししし	400000000000000000000000000000000000000	
-	,,,,,,,,,		
1 X00-10800000	00040000	10000000000000000000000000000000000000	- アシーのひららって
9			
NON-00000000		unnunnunnunnun	UNUNU44NUN
2×			
NO HOUSE	28	28	28
SZZZZZZZ OJE.	ZZZ	ZZZZZ	zzzzz
			JUULU JUUL
ALXULAGAGAGA	THE PEPE	DDDDDDDDDDDD	PADD HALL
XTAUNXXXXXXXX	XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	XXXXXMXXXXXX	XXXXMXXXXX
はいるからしませばはは	பயயியையய்ய	<u>սատաատ</u> աատաատա	THE THE PROPERTY OF THE PROPER
απ ακαακακ	KEK XXXX	xxxxxx xxxxx	****

wwww wwonnonno noomananan noomananan noomananan noomanan nooman noomanan nooman nooman nooman nooman nooman nooman nooman nooman nooman no

000001	44444444WR
•	V

			GE 03	
2,	<i>S S S S S S S S S S</i>		E I A	
ø	-000		DA 1E	<u> </u>
			366	1111111111111111
4	4400		960	>200018807440450188 01400604581459167 日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日
7.6	FIERE		0.6	4 W 7 B 7 C 7 C 7 C 7 C 7 C 7 C 7 C 7 C 7 C
4	2000		0	NOW40000011000001000000000000000000000000
r.	1060	T	" "	しているというないのできるとしてしてしているとのできるとしてしているとのできるとのできるというというというというというといいというというというというというというというと
911	900g	-	4	0000000000000000
4	98	ш	RA	
0	0000	TIGUE	MIN	00000000000000000000000000000000000000
	****	ING FA	PLASTIC	94 000000000000000000000000000000000000
•	0000	SULT	OM	00
0	0000	ES	FR	0000000000000000
-8.93	-2.64 -0.08 -0.08	NS WITH R	RAIN 06 DAMAGE	>
111111	L-10180	PLASTIC STRAIN	PLASTIC ST 0.007	Ž danaana
~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	770040 000000	AND PLA		0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
78	0000 0000	ES		×0-00000000000000000000000000000000000
XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX	319-74 P	STRESS	STEP	00000000000000000000000000000000000000
PAPAPA DAPAPA MEMUMUMUMUMUMUM MEMUMUMUMUMUMUM MEMUMUMUMUMUM MEMUMUMUMUM MEMUMUMUM MEMUMUMUM MEMUMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM MEMUM	199. 109.	LOCAL		

```
12
12
                                                                                                                                                                                                                                                        10
4.40129994.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294.40294
                                                                                                                                                                                                                                                                                          040
                                                                                                                                                                                                                                                                                   Jum
                                                                                                                                                                                                                                                                     01
                                                                                                                                                                                                                                                        014
                                                                                                                                                                                                                                                                                   mw4
                                                                                                                                                                                                                                                                                    Z 400
                                                                                                                                                                                                                                                                                         mm
                                                                                                                                                                                                                                                         3
                                                                                                                                                                                                                                                                      4
                                                                                                                                                                                                                                                                                          98
                                                                                                                                                                                                                                                         524
000
                                                                                                                                                                                                                                                                      524301
 .21
                                                                                                                                                                                                                                                          9
                                                                                                                                                                                                                                                         0
00
                                                                                                                                                                                                                                                                       9
                                                                                                                                                                                                                                                          11
                                                                                                                                                                                                                                                                                  <u>տաստարարարութարարարարարարարարարանարարարարա</u>
                                                                                                                                                                                                                                                        ET
                                                                                                                                                                                                                                                                      0
11
                                                                                                                                                                                                                                                         S
                                                                                                                                                                                                                                                        H
                                                                                                                                                                                                                                                                     CYC
 SOL
 198E
                                                                                                                                                                                                                                                                      NN
                                                                                                                                                                                                                                                         ER
                                                                                                                                                                                                                                                                      iu
                                                                                                                                                                                                                                                                                   MUN
                                                                                                                                                                                                                                                         GE
                                                                                                                                                                                                                                                                      AL
                                                                                                                                                                                                                                                         MAG
                                                                                                                                                                                                                                                                     TO
                                                                                                                                                                                                                                                                                   512
                                                                                                                                                                                                                                                                                  MOW
                                                                                                                                                                                                                                                                                  29. R
                                                                                                                                                                                                                                                         DA
+
                                                                                                                                                                                                                                                                                   Z
                                                                                      ユーーーーーーークシンクシンクシンクシック
                                                                                                                                                                                                                                                                                  200000
244
200000
                                                                                                                                                                                                                         4900
                                                                                                                                                                                                                                                                                           NUNNN
                                                                                                                                                                                                                                                                                           11111
                                                                                                                                                                                                                                                                               000 W 
NC0000
  11111111111111
                                                                                                                                                                                                                                                                            SZ www
                                                                                                                                                                                                                                                                                                        ...
3.22
3.22
3.22
                                                                                                                                                                                                                                                                                                      777
                                                                                                                                                                                                                                                                                 S
                                                                                                                                                                                                                                                                                                      TAMMAXXX
                                                                                                                                                                                                                                                                           DAD . . 20
                                                                                                                                                                                                                                                                           しているかししし
```

20008765	2000-00-00-00-00-00-00-00-00-00-00-00-00						
	7	55	32	45			
	n	4	'n	•			
	5	40	40	40			
	7	C7E	0 3E	47E			
2,000,0	76	295	053	955			
	7	0 2	73	.4719			
		0.9	0.3	4.0			
444444		00000000000	* 000000000 * 000000000 * 000000000 * 0000000000				
		00	0.	0			
2	Î	51	o	o			
5		92	8 9	93			
a	•	. 56.	9.00	8			
		7	7	•			
4444444	00000000000000000000000000000000000000	00-10mm44m4 04-10m-1-10m	24400000000000000000000000000000000000	100000mm			
******* a		28	8 8	7			
PAPAPA	XXXXXXXXXX PPPPPPPP UUUUUUUUUUUUUU	TITITITITITITITITITITITITITITITITITITI	ZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZZ	ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXATICN ELAXAT			

2000	
0.61621797E 04 7 52 0.83860000E C4 8 53 0.12079809E C5 9 54 0.18907535E C5 10 55	NIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN STRAIN S
***** ***** ***** ***** **** **** ****	RESULT IN G
5.33 6.11 6.28 8.43 -0.49 8.90 -0.08	LASTIC STRAINS WITH  LASTIC STRAIN  DO 00706  DAMAGE  RNCYC  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.  155.
284621 284621 284621 38622 38622 38632	AND PLAST I
RELAXATION RELAXATION RELAXATION 16.39 3.28 13.41 3.28 10.43 3.28	S. I. S.

**ららってことととととととととしてしてしてしてい** 4001

արաաաաաաաաաաաաաաաաաաաա

2425 8E 8418 9 0 7 ET 0 S THIS ENN/CY ER 0 AL MAGE TO

らっしらしてっていることなっているとのののの 0-4-0000-0mm00000-0000mm 0/m4700000/m44700000000400 AIWAIAIAIAIWWAIAIWWAIAIRIWWAIAIAUWAIAIAUW

## INPLT OF RANCU GENERATED LOADS

016680	.2801 .5154	683	1	
A7 CRITICAL PCINT #1 FRGM	1.002	100000001	-4	1 1 29.8

GUTPLT CF 4201 RANDU GENERATED LOADS, IPRINT =

7)

SPECTRUM SUBJECTED TO THE RANGE-PAIR CCLNTING TECHNIQUE SEQUENCE ACCCUNTABLE FATIGUE EVALUATION

					:	70000 70000 29595 55999 56999			1.000			m
	3	,			N DATA	717 344 38			E2 =			
#1 FRCM U166808	72.00000	0.40000	-1.83600	0	WI.	0.28010 0.51540 0.61410 0.68380	FROM K-5A	CN	EAN**E2) E1 = 1.000 AND	4201 LOAES	6	N F000000000000000000000000000000000000
A7 CRITICAL POINT	STRESS (KSI)	INTERCEPT = 0	COFFIN-MANSON SLOPE	.US = 10000.00000	CF SECOND ORDER LEAS I)*SMIN**2 + B(I)*SM	-0.00200 -0.00220 -0.00140 -0.00130	COUPON S-N DATA DERIVED	SS RELAXATION FUNCTI	/(KTSMAX**E1 * KTSM C.10000000E 08 *	IES THROUGH BLOCK OF	IIT = 29.79999	NT
SPECTRUM FRCM	NSILE YIE	LCF STRAIN IN	INVERSE CF CO	ELASTIC MCDULUS	TXL	10 * * * * * 10 * * * * * * * * * * * *	UNNOTCHED COU	RESICUAL STRE	WHERE CI =	1 TIM	LCAC LIM	22222222222222222222222222222222222222

0.35000 0.35000 0.35000	TYPE
	TYPE
	TYPE
000	TYPE
0.11000	TYPE
	TYPE
	TYPE
<del></del>	1YPE
3701 3951 4201	BLOCK

SPECTFUM FRCM A7 CRITICAL POINT #1 FROM U166808

IKT = 2.7

RELAXATION CONSTANT CI= 40000000.00

DAMAGE FROM PLASTIC STRAINS= 0.46892278E-03 SPECTRUM SUBJECTED TO THE RANGE-PAIR CCUNTING TECHNIQUE FLIGHT CR BLOCK NO. 1

DAMAGE PER THIS SET= 0.46267733E-C2

TCTAL ENN/CYC =, 0.46267733E-02

## INPUT OF RANGE GENERATED LOADS

U16689g	 1000	1
FROM		
A7 CRITICAL PCINT #1	000	4.

71.7MIL-HCBK-5A 56.3MIL-HDBK-5A 44.6MIL-HDBK-5A 38.1MIL-HCBK-5A 1. OUTPUT CF 4201 RANDU GENERATED LOADS, IPRINT = 3

1 CATA DECKS ARE TO BE PROCESSED.
NO CCUNTING METHODS USED
SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

						-0555 1-0555 1-05555 1-05555			1.000			M
					S-N DATA	2450 1048 1048			: E2 =			
0166808	72.00000		1.83600		FIT OF	(1) 8010 11540 8380			1.000 ANE	1 LOADS		E0000000000000000000000000000000000000
#1 FRCM	72	0.40000	1	000	ST SQUARE	0000	D FROM	ICN	SMEAN**E2)	4201	666	
CRITICAL POINT	(KSI)		SON SLOPE	10000.00000	R LEA	-0.00200 -0.00220 -0.00140 -0.00130	DATA DERIVER	TION FUNCTI	E1 * KT	SH BLOCK OF	29.19999	XI
TAI	STRESS	STRAIN INTERCEPT =	COFFIN-MANSON	= 507	(I) *SM	40000	GRMATICN SUPPLIED F	STRESS RELAXATION	C1/(KTSMAX**	TIMES THROUGH	LIMIT =	
CTRUN FRCM	SILE YIELD		INVERSE OF C	ELASTIC MODULUS	FF	# * * * * 10	CTCHEC CO	ICUAL STR	ENEP = C	1 11	LCAC LI	
SPE	TEN	LCF	INV	ELA	COE	•••	INF	RES	WHE			8 111110202020 E 140202020 E 1402020202020202020202020202020202020202

	TYPE	TYPE	TYPE	TYPE	TYPE	TYPE	TYPE	TYPE	BLOCK
11.	0.35000			110000					4201

SPECTRUM FRCM AT CRITICAL PCINT #1 FRCM U166808

AKT = 2.7

RELAXATION CONSTANT C1= 10000000.00 FLIGHT OR BLOCK NO. 1

DAMAGE FROM PLASTIC STRAINS= 0.46892278E-03

DAMAGE PER THIS SET= 0.47398917E-02

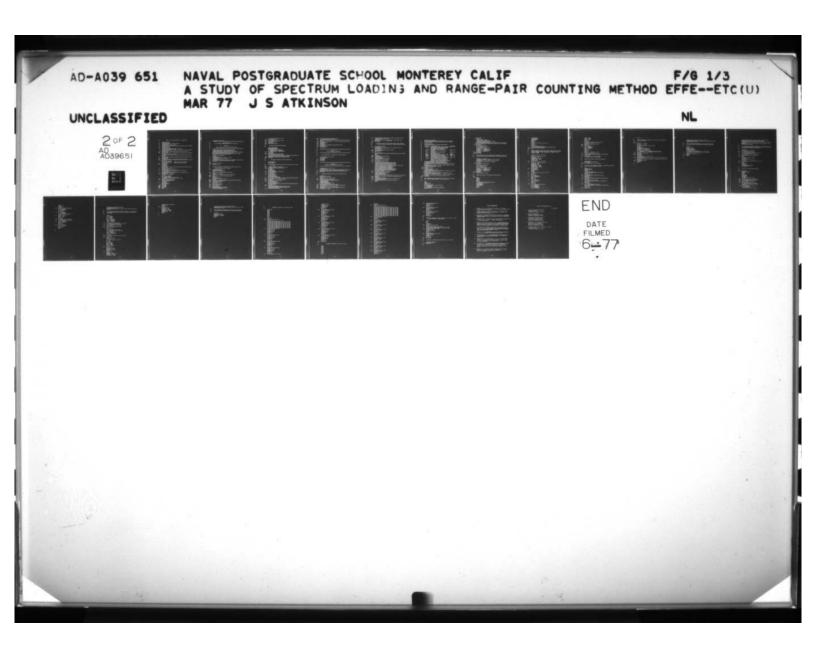
TCTAL ENN/CYC =, 0.47358517E-02

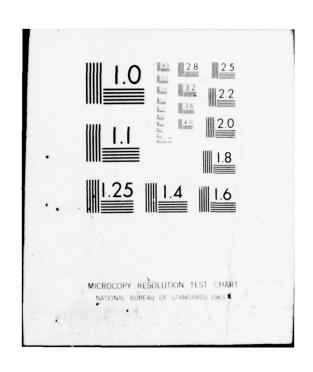
## COMPUTER PROGRAM

*****	********
SEQUENCE AC	MODULE I UT ROUTINE FOR THE COUNTABLE FATIGUE ANALYSIS
INPUT	
DATA CARD 1.	NDECK = THE NUMBER OF DATA CECK BE RUN SEQUENTIALLY THE VALUE CONTROLLING TO WRITE STATEMENTS 1. PERMITS MAXIMUM PRINT 2. SUPPRESSES RANGE-PAIR PRINTING 3. MAXIMUM SUPPRESSION
	IRPCM = THE VALUE CONTROLLING TO ENTRY INTO THE RANGE-PA COUNTING SUBROUTINE  1. ENTER RANGE-PAIR COUNTINE 2. SKIP RANGE-PAIR COUNTINE 2. SKIP RANGE-PAIR COUNTINE 314
EACH DATA DECK CO	NTAINS THE FOLLOWING CARES -
CARD 1.	TEST IDENTIFYING INFORMATION FORMAT 1644
CARC 2.	TM = MATERIAL TYPE TYS = TENSILE YIELD STRESS (K EPSD = LCF STRAIN INTERCEPT COFMAN =INVERSE CF COFFIN-MANSO SLOPE ELMOD = MODULUS OF ELASTICITY ( FORMAT 4A4,3F18.5,F10.2
CARCS 3,,6.	A(N) N=4,7 (A,B,C ARE CCEFFIC OF SECOND CROER LE SQUARE FIT OF S-N FOR CURVE OF 10**N CYCLES.)
	(STMAX = A(N)*STMI + B(N)*STMIN + HELL, TITLE2, TIT TITLE2 TITLE4 IDENTIFIES TITLE3 SCURCE OF THE SOURCE OF THE TITLE4 IN COLUMN FOR INFO.) MATERIAL TYPE (COLUMNS 7380. INFO ONLY.)
	FORMAT 3F18.5,444
CARD 7.	C1 (CONSTANTS TO BE LSED II CALCULATION OF EQUILIBR PERIOD, ENEP.) (ENEP=C1/(KTSMAX**E1*KT **E2)) FORMAT 3F18.5

OOOOL	CARC 8.	AKT = STRESS CONCENTRATION FACTOR USED THE FIRST TIME THROUGH THE PROGRAM F18.5
,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	CARD 9.	NBLOCK = NUMBER OF BLOCKS (NO. OF TIMES TO REPEAT LIST OF LOADS)  NLEVEL = NUMBER OF LOADS  NTYPE = NUMBER OF TYPES OF LOADS  FORMAT 3110
Č C C	CARD 10.	TLL = LIMIT LCAD (KSI) FORMAT F18.5
งบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบบ	CARDS 11,,	NLEVEL + 10.  K THE KTH STEP (K=1, NLEVEL)  ITYPE(K) = IDENTIFYING TYPE OF THE KTH LOAD  STMIN(K) = THE KTH MINIMUM (DECIMAL FRACTION OF TLL)  STMAX(K) = THE KTH MAXIMUM (DECIMAL FRACTION OF TLL)  ENN(K) = NUMBER OF CYCLES AT THE KTH LOAD  FORMAT 2(14,2X),3(F18.5,1X)
ดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดดด	CARDS NLEVEL + 11	THE JJTH BLOCK (JJ=1,NBLCCK)  NN(JJ,1) = TYPE CF LOAD INCLUDED IN  JJTH BLOCK  NN(JJ,2) = (THERE WILL BE ONE NN  VALUE ON THE CARC FOR  NN(JJ, ) = EACH CIFFERENT TYPE OF  LOAD INCLUDED IN THE JJTH  BLOCK)  FORMAT 915

```
CCMMCN/MSAL/RNCYC ( 4205), KPMAX, IPRINT
CIMENSION RTMIN(4205), RTMAX(4205), ITYPE(4205), RNN(4205
               1)
CCMMON/MCGRI/NLEVEL, IRPCM, I FLOT, ELMOD, TYS, EFSC, COFMAN,
1RES( 4205), AKT, SUMENN, SUMNC, C1, E2, E1
CCMMON/MCGR2/STMIN( 4205), STMAX( 4205), ENN( 4205),
1C(10), R(10), NN(20,10), A(10), B(10)
CIMENSICN HOLD(101)
              INSERT RANDU PACKET HERE - REMOVE ISKIP=0 CARD
ISKIP=0
REAC(5,4) NDECK, IPRINT, IRPCM
FCFMAT(314)
WRITE(6,3) NDECK
FCRMAT(1H1, I4,32H DATA DECKS ARE TO BE PROCESSED.)
IF (IRPCM.GE.2) GG TO 6
WRITE(6,11)
GC TO 13
WRITE(6,7)
FCRMAT(25H NO COUNTING METHODS USED)
CCNTINUE
DC 595 LMN = 1, NDECK
WRITE(6,5)
FCRMAT(40H SEQUENCE ACCCUNTABLE FATIGUE EVALUATION)
READ(5,9)T1,T2,T3,T4,T5,T6,T7,T8,T9,T10,T11,T12,T13,T1
14,T15,T16
FCFMAT(16A4)
WRITE(6,8)T1,T2,T3,T4,T5,T6,T7,T8,T9,T10,T11,T12,T13,T
114,T15,T16
FCFMAT(16HISPECTRUM FRCM ,16A4)
4
3
67
13
5
9
800000
                                          INPUT OF DATA PECULIAR TO A MATERIAL
              10
12
14
16
18
20
22
56
24
26
28
```





```
INPUT OF DATA PECULIAR TC A SEQUENCE
65
32
34
35
33
89
51
55
11
59
                  DC 1002 KFL=1,NBLOCK

JJJ=1
CC 70 KK=1,NTYPE
IF (NN(KFL,KK).EQ.0) GO TO 60
IF (ITYPE(J).EQ.NN(KFL,KK)) GO TO 150
CCNTINUE
STMIN(JJJ)=RTMIN(J)*TLL
STMAX(JJJ)=RTMAX(J)*TLL
ENN(JJJ)=RNN(J)
JJ=JJJ+1
CCNTINUE
CLATINUE
CLATINUE
CCNTINUE
60
1002
597
595
596
580
C
```

```
SLERGUTINE CORE(KFL)
   COCOCOCOCOCO
                                    THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS
                                                                                          MODULE II
                                           LOCAL STRESS AND STRAIN DETERMINATION
                  CCMMCN/MCEC1/SIGMAX( 4205), SIGMIN( 4205)
CCMMCN/MSAL/RNCYC( 4205), KPMAX, IPRINT
DIMENSICN PLSTRA( 4205), EN( 4205), EX( 4205)
CCMMCN/MCOR1/NLEVEL, IRPCM, IPLOT, ELMOD, TYS, EFSC, COFMAN,
1RES( 4205), AKT, SUMENN, SUMNC, C1, E2, E1
CCMMCN/MCGR2/STMIN( 4205), STMAX( 4205), ENN( 4205),
1C(10), R(10), NN(20,10), A(10), B(10)

JJ=KFL
IF (JJ.GT.3) IPRINT=3
IRAIN=1
HRITE(6,54) JJ
FCFMAT(20H FLIGHT OR BLCCK NO., I5)
IF (IPRINT.GE.3) GO TO 61
WFITE(6,62)
FCRMAT(1X,6H STMAX,3X,5HSTMIN,3X,6HSIGMAX,3X,6HSIGMIN,
13X,3HRES,3X,5HEQRES,3X,3HENN,11X,3HNEP)
CCATINUE
   54
   62
  61
C
148
149
                      OC 570 J=1, NLEVEL I=J+1 PLSTRA(J)=0.0
0000000
                       CETERMINE SEQUENCE ACCOUNTABLE RESIDUAL STRESS AND
                                    CORRESPONDING PLASTIC STRAIN
                       ASMAX=AKT+STMAX(J)
ASMIN=AKT+STMIN(J)
ASMEAN=(ASMAX+ASMIN)/2.
                       JA=C

IF (RES(I-1)+ASMIN+TYS) 190,190,200

JA=1

AAA=RES(I-1)+ASMIN

FLSTRA(J) = -1.*(AAA/ELMOD)*(1.+AAA/TYS)
   170
                       JE=0
IF (RES(I-1)+ASMAX-TYS) 220,210,210
                     IF (RES(I-1)+ASMAX-TYS) 220,210,210

JB=-1

AAA=ASMAX

BEB=TYS-RES(I-1)

IF (BBB-GE-TYS) GC TO 214

BEB=0.0

FLSTRA(J) = AAA*AAA/(ELMOD*TYS)-BEB*BBB/(ELMOC*TYS)

IF (JA+JB) 230,250,240

RES(I)=TYS-ASMAX

GC TO 290

RES(I)=-TYS-ASMIN

GC TO 290

RES(I)=RES(I-1)

GC TO 290

RES(I)=RES(I-1)

GC TO 290

RES(I)=-ASMEAN

SIGMAX(IRAIN)=RES(I)+ASMAX

SIGMIN(IRAIN)=RES(I)+ASMIN

RNCYC(IRAIN)=ENN(J)
  212
214
220
230
   240
```

```
IF (ASMAX.LE.TYS) GO TO 410

ECRES=-ASMAX+TYS

GC TO 440

IF (ASMIN.GE.-TYS) GO TO 430

ECRES=-ASMIN-TYS

GC TO 440

ECRES=0.0

DIF=RES(I)-EQRES
 400
 410
4400000
                                              CALCULATE RELAXATION FUNCTION
                           ABMAX=ABS(ASMAX)
ABMIN=ABS(ASMIN)
ABMEAN=ABS(ASMEAN)
IF (ABMAX.LT.1.) ABMEAN=0.5
IF (ABMIN-ABMAX) 444,444,442
AEM=ABMIN
GC TO 446
AEM=ABMIN
ENEM=CI/(ABM**E1*ABMEAN**E2)
IF (IPRINT.GE.3) GO TO 351
WRITE(6,350) STMAX(J),STMIN(J),SIGMAX(IRAIN),SIGMIN(IRAIN),RES(J),EQRES,ENN(J),ENEP,J,IRAIN
FCRMAT(6(F7.2,1X),F6.2,1X,E15.8,I5,I5)
CCNTINUE
 442
 444
350
351
CCCCC
                                                               CALCULATE RESIDUAL STRESS RELAXATION
                                   IRAIN=IRAIN+1
AECIF=ABS(CIF)
GC TO 360
CCC 360
                                  CMITTED "GO TO 560" CARD FROM THIS POSITION, AS SHOWN IN MANUAL
                               IF (ABDIF.LT.5.) GO TO 560

IF (1000.*ENN(J).LT.ENEP) GO TO 56C

IF (ENN(J).LE.10.) GO TO 560

NFLAG=0
NFLAG=0
NFLAG=10
IRAIN=IRAIN-1
CUMMY=ENN(J)
IF (DUMMY-ENEP) 470,460,460
CLMMY=DUMMY/2.
NFLAG=NFLAG+1
GC TO 450
CYCINT=DUMMY/10.
CC 500 K=1:0
CECK=FLOAT(K)
EN (K)=CYCINT*DECK
IF (K.EC.1) GO TO 490
EX(K)=EXP(-2.303*EN(K-1)/ENEP)+EXP(-2.303*EN(K)/ENEP)
GC TO 500
EX(K)=1.+EXP(-2.303*EN(K)/ENEP)
CCNTINUE
IF (NFLAG2=NFLAG+1)
CCNTINUE
IF (NFLAG2=NFLAG+1)
CCNTINUE
CX (K)=2.*DUMMY
CX (IRAIN)=ENGMAX+ECRES+DIF*EX(K)/2.
SIGMAX (IRAIN)=ASMAX+ECRES+DIF*EX(K)/2.
SIGMAX (IRAIN)=ENGMAX (IRAIN)-ASMAX+ASMIN
RNCYC(IRAIN)=ENGMAX (IRAIN)-ASMAX+ASMIN
RNCYC(IRAIN)=ENGMAX (IRAIN)-ASMAX+ASMIN
RNCYC(IRAIN)=ENGMAX (IRAIN)-ASMAX+ASMIN
IF (K.EG.1) GO TO 540
 380
 450
470
 480
 490
 510
 520
 545
```

```
RNCYC(IRAIN)=RNCYC(IRAIN)-EN(K-1)

IF (IPRINT.GE.3) GO TO 551

WRITE(6,550)SIGMAX(IRAIN),SIGMIN(IRAIN),RNCYC(IRAIN),I

IRAIN
FCRMAT(16H RELAXATION ,2(F7.2,1X),16X,F6.2,31X,I6)

CCNTINUE
IRAIN=IRAIN+1
CCNTINUE
CCNTINUE
RES(I)=EQRES+DIF*EXP(-2.303*ENN(J)/ENEP)
CCNTINUE
RES(I)=RES(J)
IN=IRAIN-1
 539
 550
551
 559
 570
0000000000000
                                                       CYCLE COUNTING TECHNIQUE
                         CALL SUBROUTINE TO RANGE-PAIR COUNT SPECTRUM
                  IF(IRPCM.GT.1) GO TO 591
CALL RPCM(IN)
GC TO 592
CCNTINUE
KFMAX=IN
CCNTINUE
 591
 592
 DAMAGE ACCUMULATION CALCULATION
                IF(IPRINT.GE.3) GO TO 552

#FITE(6,53)

FCRMAT(//1x,39H LGCAL STRESSES AND PLASTIC STRAINS WIT
124HH RESULTING FATIGUE LIFE//10x,4HSTEP,10x,
114+PLASTIC STRAIN,10x,10HMAX OR MIN,15x,6HDAMAGE)

CCNTINUE
 53
552
CCCCC
                            CALCULATE DAMAGE FROM PLASTIC STRAIN CYCLES
                   SUMDEL = 0.
CC 531 JKL=1, NLEVEL
                    AA=1.
IF(PLSTRA(JKL)) 532,531,533
                 IF(PLSTRA(JKL)) 532,531,533

AA=-1.

PLSTRA(JKL)=AA*PLSTRA(JKL)

CYCLES=(PLSTRA(JKL)/EPSD)**CCFMAN

CAP=1./CYCLES

SLMNC=SUMNC+DAM

SLMDEL=SUMDEL+DAM

IF(IPRINT.GE.3) GD TO 531

IF (AA) 535,535,537

WRITE(6,199)JKL,PLSTRA(JKL),CAM

FCRMAT(10X,I4,12X,F10.5,15X,3HMIN,10X,E14.6)

GC TO 531

WFITE(6,219)JKL,PLSTRA(JKL),CAM

FCRMAT(10X,I4,12X,F10.5,15X,3HMAX,10X,E14.6)

CCNTINUE

WFITE(6,541) SUMDEL
532
 535
```

```
FCRMAT(35X,29H DAMAGE FROM PLASTIC STRAINS=,E15.8)
IF (IPRINT.GE.3) GO TO 536
hRITE(6,13)
FCRMAT(, 8x,15H SIGMAX SIGMIN, 8x,6H RNCYC, 8x,
125H CYCLES
CONTINUE

FCRMAT(35X,29H DAMAGE FROM PLASTIC STRAINS=,E15.8)

REPROPERTY OF THE PROPERTY OF THE PROPERTY
             541
           13
536
                                                                                                                                                                                  CALCULATE ELASTIC CYCLE DAMAGE FROM LEAST SCUARE FITTED S-N DATA (MODIFIED GOODMAN DIAGRAM FORMAT)
                                                                                                     DC 600 JKL=1, KPMAX
T1YS=1YS/5.
IF (SIGMAX(JKL)-SIGMIN(JKL).LT.1.6*TYS) GO TC 310
CYCLES=10.**4.
GC TO 340
IF (SIGMAX(JKL).GE.TTYS) GO TO 320
CYCLES=10.**4.
GC TO 340
L=1, KPMAX

L=1, KP
                534
             320
           335
           336
             338
             339
           340
           599
             593
             575
           C
```

```
SLEROUTINE RPCM(NPKS)
THIS PROGRAM EMPLCYS THE RANGE-PAIR CYCLE CCUNTING METHOD TO GENERATE ANALYSIS SPECTRUM FROM A GIVEN LCAC SPECTRUM
ひいしついっしついっしついっしついっしついっしついっしつ
                                                             PECGRAM ARRAYS (INFORMATION NEEDED TO CHANGE DIMENSIONS)
                      PREGRAM ARRAYS
(INFORMATION NEEDED TO CHANGE DIMENSIONS)

ARRAY NAME
SIGMAX

PEAKS OF THE INPUT LOAD SPECTRUM NPKS + KK

THE NUMBER OF ADDITIONAL CYCLES
(EXCLUDING INPUT CYCLES) WHICH

THE PROGRAM WILL GENERATE

SIGMIN VALLEYS OF THE INPUT LOAD NPKS + KK

RNCYC K COUNTERS OF THE PEAKS AND NPKS + KK

ASTEP STEP NUMBERS OF THE INPUT NPKS + KK

RES RESIDUE SPECTRUM

RES RESIDUE SPECTRUM

RES RESIDUE SPECTRUM

RES RESIDUE SPECTRUM

RNCYC K COUNTERS OF THE EMENTS IN RES 2*NPKS

CYCLE RANGE-PAIR COUNTED CYCLES OF THE NPKS + KK

NNSTEP STEP NUMBERS OF THE ELEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE ELEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE ELEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE ELEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE LEMENTS OF NPKS + KK

NNSTEP STEP NUMBERS OF THE INPUT

NOTH ON NPKS + KK

NNSTEP STEP NUMBERS OF THE NPWT

NOTH ON NPKS + KK

NNSTEP STEP NUMBERS OF THE NPWT

NOTH
          9999
           80CC
                                                             SCRT THROUGH THE LOAD SPECTRUM - PULL OUT THOSE PEAKS AND VALLEYS WHERE COUNTER K IS LESS THAN 1.0
                                                           J=1

L=C

ARES = 1

ACYNO = 100

JFAX = 0

CC 100 I=1, NPKS

IF (RNCYC(I) .GE. 1.0) GO TO 100

X1 = SIGMAX(I)

X2 = SIGMIN(I)

CALL CYCGEN(X1, X2, RNCYC(I), NSTEP(I))

ISAVE(J) = I
```

```
COCOCO
                                SCRT THROUGH THE LOAD SPECTRUM DATA-COMBINE STEPS WITH ICENTICAL PEAKS AND VALLEYS WHICH CCCUR CONSECUTIVELY
                DC 300 I = 2, NPKSN

IF (SIGMAX(I) · NE· SIGMAX(I-1)) GC TO 300

IF (SIGMIN(I) · NE· SIGMIN(I-1)) GC TO 300

IF (SIGMIN(I) · NE· SIGMIN(I-1)) GC TO 300

IF (SIGMIN(I) · NE· SIGMIN(II-1)) GC TO 300

IF (J·EC·I) GO TO 6000

JF (J·EC·I) GO TO 6000

JF (J·EC·I) GO TO 6000

JF (J·EC·I) GO TO 311

CC 311 J = 1, JMAS

I = ISAVE(J) - (J-I)

NPKN = NPKSN - J

IF (I·EC·NPKN) GO TO 311

CC 316 II = I, NPKN

SIGMAX(II) = SIGMAX(II+1)

SIGMIN(II) = SIGMIN(II+1)

NSTEP(II) = NSTEP(II+1)

RCCYC(III) = RNCYC(II+1)

11 CCNTINUE

NFKSN = NPKSN - JMAS
           00000
                               RANGE PAIR COUNT THE ADJUSTED LOAD SPECTRUM
                               K=1=0

IF (RNCYC(I) .GT. 1.0) GO TO 400

IF (KB .NE. 0) GO TO 5

X1=SIGMAX(I)

X2=SIGMIN(I)

IND1=NSTEP(I)

INC2=IND1

I=I+1

KB=1

GC TO 1
           1
```

```
x3=SIGMAX(I)

x4=SIGMIN(I)

IND3=NSTEP(I)

INC4=IND3

KMIN=I

KMAX=0

K31=0

IF (RNCYC(I) .EQ. 1.0) GO TO 6

KEY=1

GC TO 415

KEY=0

CYCNC=AINT(RNCYC(I)+0.5)

CALL CECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)

GC TO (10,10,30),KCYGEN

KE=1
     5
     6
100 C
                                                                                                    REVERSE CRDER OF NEXT TWO CARDS TO RUN. "I=I+1" GCES AFTER "IF(KMIN.NE.1) GO TO 36" AS PER ERRATA SHEET IN FRINTED MANUAL. CHECK PROGRAM WITH CARDS IN THE MANUAL'S CROER
                                                                                           MANUAL'S CROER

IF (KMIN .NE. 1) GO TO 36

I=I+1
IF (I .LE. NPKSN) GO TO 5

RES(LR+1) = X1
RES(LR+2) = X2
INDEX(LR+2) = IND2
IRES(LR+2) = IND2
IRES(LR+2) = X2
GC TO 2000
IF (KMIN .NE. 1) GO TO 35
I=I+1
IF (I .LE. NPKSN) GO TO 31
RES(LR+2) = X2
RES(LR+2) = X2
RES(LR+2) = X3
INDEX(LR+2) = IND3
IRES(LR+2) = IND3
INDEX(LR+3) = IND3
INDEX(
 31
 32
 40
 35
 400
```

```
IND2 = IND1
IND3 = IND1
IND4 = IND1
KMIN = 1
KMAX = 0
K31 = 0
IF (RNCYC(I) = RNCYC(I) - 1.0
GC TO 402
RNCYC(I) = RNCYC(I) - 2.0
KIND = 0
GC TO 415
X3 = SIGMAX(I)
X4 = SIGMIN(I)
IND3 = NSTEP(I)
INC4 = IND3
KMIN = 1
KMAX = 0
K31 = 0
KIND = 0
CYCNO = AINT(RNCYC(I) + 0.5)
CALL DECIDE(X1, X2, X3, X4, KEY, I, CYCNO, KCYGEN)
GC TO 100
LMAX = L
IF (LRMAX .LT. 4) GO TO 5000
IF (NCYNO .EQ. 0) GO TO 5000
410
415
200C
                            RANGE PAIR COUNT OF RESIDUE SPECTRUMS
                          NRES=NRES+1
CALL DECRES(LRMAX, NCYNO)
GC TO 2COO
IF (LRMAX .LE. 1) GO TO 3000
                            CCUNT THE LAST RESIDUE SPECTRUM - RANGE-PAIR COUNTING WILL YIELD N ADDITIONAL CYCLES
                          KK=0
RESMAX=RES(1)
RESMIN=RES(1)
IMAX=1
IMIN=1
CC 500 I=2,LRMAX
IF (RES(I) .LT. RESMAX) GO TO 490
RESMAX = RES(I)
IMAX = I
GC TO 500
IF (RES(I) .GT. RESMIN) GO TO 500
RESMIN=RES(I)
IMIN = I
CCATINUE
CALL CYCRES(RESMAX,RESMIN,1.0,INCEX(IMAX))
KK = KK+1
J=IMAX-2
IF (J .LE. 0) GO TO 550
CALL CYCRES(RES(J),RES(J+1),1.0,INDEX(J))
KK = KK+1
IMAX = J
GC TO 510
J = IMIN + 2
IF (J .GT. LRMAX) GO TO 575
CALL CYCRES(RES(J-1),RES(J),1.0,INDEX(J-1))
KK = KK+1
IMIN = J
GC TO 550
KMAX = KK
490
500
510
550
575
```

```
SLBROUTINE DECIDE(X1, X2, X3, X4, KEY, I, CYCNC, KCYGEN)
CCMMON/MDEC1/SIGMAX( 4205), SIGMIN( 4205)
CCMMON/MDEC2/NSTEP( 4205), LR, KMAX, KMIN, K31
CCMMCN/MDECR/RES( 4205), INDEX( 4205), IND1, IND2, IND3, IN
1C4, KIND
CCMMCN/MCYG/CYCLE(200, 2), RNECYC( 4205), NNSTEP( 4205)
CCMMON/MCGDE/L, LIND
                                 THIS SUBROUTINE DECIDES WHETHER OR NOT THE VALUES X1, X2, X3, AND X4 FROM THE ADJUSTED LOAD SPECTRUM SATISFY THE RANGE-PAIR COUNTING CONCITIONS
                                KFIRST=0
IF (K31.NE.0) GO TO 11
IF (X3.LE.X2) GO TO 200
IF (X2.GT.X1) GO TO 210
IF (X2.GT.X3) GO TO 210
IF (X2.GT.X3) GO TO 151
CALL CYCGEN(X3,X2,1.0,NSTEP(I))
GC TO 152
CALL CYCGEN(X2,X3,1.0,NSTEP(I))
X1=X1
X2=X4
IF (INC3.NE.IND2) LIND=1
INC2=INC4
KCYGEN = 1
IF (KEY.NE.0) GO TO 110
RETURN
IF (X2.GT.X4.OR.X3.LT.X1) GO TO 500
GC TO 150
X1=X1
 10
  150
 210
                                CC TO 150

X1=X1

X2=X4

INC2=IND4

KCYGEN = 2

IF (KEY.EQ.O) RETURN

CYCNG=CYCNO-1.0

GC TO 110
  200
000005
                                 ACC X1 TO THE RESIDUE SPECTRUM
                              LR=LR+1
RES(LR)=X1
INDEX(LR)=IND1
X1=X2
X2=X3
X3=X4
IND1=IND2
IND2=IND3
INC3=INC4
KCYGEN = 3
IF (KEY.NE.O) GO TO 110
RETURN
GC TO (1150,1200,1500),KCYGEN
IF (CYCNO.GT.1.O) GO TO 1151
IF (CYCNO.GT.1.O) RETURN
CYCNC = CYCNO - 1.0
CYCNC = CYCNO - 1.0
IF (LINC.EQ.1) GO TO 1153
IF (INC3.NE.IND4) GO TO 1153
IF (INC3.NE.IND4) GO TO 1253
RNECYC(L) = RNECYC(L) + CYCNO - 2.0
CYCNC = 1.0
IF (KMAX.NE.1) GO TO 111
X3 = SIGMIN(I)
IND3 = NSTEP(I)
IF (CYCNO.GT.0.0) GO TO 112
 1100
 1153
 1151
 1152
```

```
KMIN = 1

KMAX = 0

KCYGEN = 3

FETURN

1200 IF (CYCNO.LE.O.O) RETURN

CYCNO = CYCNO - 1.0

X3 = SIGMAX(I)

X4 = SIGMIN(I)

KFIRST = 1

GC TO 113

111 X3 = SIGMAX(I)

IF (KFIRST.NE.O) GO TO 113

CYCNO = CYCNO - 1.0

KFIRST = 1

113 INC3 = NSTEP(I)

IND4 = IND3

KMIN = 1

KMAX = 0

GC TO 10

1500 IF (KMAX.NE.O) GO TO 1510

IF (CYCNO.LE.O.O) RETURN

CYCNO = CYCNO - 1.0

112 X4 = SIGMAX(I)

IND4 = NSTEP(I)

KMAX = 1

KMIN = 0

GC TO 11

IND4 = NSTEP(I)

KMAX = 0

KMIN = 0

GC TO 10

INC4 = NSTEP(I)

KMAX = 0

KMIN = 1

KMIN = 0

GC TO 10

CCCC
```

```
SLERCUTINE DECRES(LRMAX, NCYNO)
CCMMON/MCGDE/L, LIND
CCMMCN/MDECR/RES( 4205), INDEX( 4205), IND1, IND2, IND3, IN
104, KIND
                                       THIS SUBROUTINE DECIDES WHETHER CR NOT THE ELEMENTS CF THE RESIDUE SPECTRUM SATISFY THE RANGE PAIR-CCUNTING CONDITIONS
                                       K = 0

NCYNO = 0

X1 = RES(1)

X2 = RES(2)

X3 = RES(3)

X4 = RES(4)

IND1 = INDEX(1)

INC2 = INDEX(3)

IND3 = INDEX(4)

INC4 = INDEX(4)
                                   INC2 = INDEX(3)
IND3 = INDEX(3)
INC4 = INDEX(3)
INC4 = INDEX(3)
INC4 = INDEX(3)
INC4 = INDEX(4)
IF (X2.GT.X1) GO TO 100
IF (X2.GT.X3) GO TO 50
CALL CYCRES(X3,X2,1.0,IND3)
GC TO 152
CALL CYCRES(X2,X3,1.0,IND2)
NCYNO = NCYNO + 1
X1 = X1
INC2 = IND4
IF (J.EQ.LRMAX) GO TO 300
IF ((J+1).EQ.LRMAX) GO TO 315
X3 = RES(J+1)
IND3 = INDEX(J+2)
IND3 = INDEX(J+2)
IND4 = INDEX(J+2)
JGC TO 10
IF (X2.GT.X4.OR.X3.LT.X1) GO TO 500
GC TO 150
K = K + 1
RES(K) = X1
INDEX(K) = IND1
J = J+1
INDEX(K) = IND1
J = J+1
INDEX(K) = IND3
INC3 = IND4
IND4 = IND5
IND2 = IND5
IND2 = IND4
IND4 = IND5
IND2 = IND5
IND3 = IND4
IND4 = IND5
IND5 = IND6
IND6X(K+1) = IND2
INDEX(K+1) = IND2
 10
 150
151
100
500
300
315
```

```
INDEX(K+2) = INDEX(J+1)

LRMAX = K+2

RETURN

K = K+1

RES(K) = X2

RES(K+1) = X3

RES(K+2) = X4

INDEX(K) = IND2

INDEX(K+1) = IND3

INCEX(K+2) = IND4

LRMAX = K+2

RETURN
END

CC
CC
```

SLERCUTINE CYCRES(Y1, Y2, CYCPF, NSTEPP) CCMMGN/MCYG/CYCLE(200,2), RNECYC( 4205), NNSTEP( 4205) CCMMGN/MCGDE/L, LIND

THIS SUBROUTINE GENERATES CYCLES FOR THE ANALYSIS SPECTRUM FROM DATA SUPPLIED BY SUBROUTINE DECRES

L = L+1 CYCLE(L,1) = Y1 CYCLE(L,2) = Y2 RNECYC(L) = CYCPF NNSTEP(L) = NSTEPP RETURN END

```
RANDOMIZE SPECTRUM A LOACS FOR STMAX
     700
                                                 751234567
755577557
75577557
                                          T000000000000000
750
751
753
```

```
ITCTAL=ITCTAL+1
IE=IE+1
RTMAX(I)=.85
RTMIN(I)=.11
ITYPE(I)=1
RTNN(I)=1.
I=I+1
GC TTO TCO
IF(ITCTAL=ITCTAL+1
RTMAX(I)=.75
RTMIN(I)=.11
ITYPE(I)=1
RNN(I)=1.
I=I+1
GC TTO TCO
IF(IG.EC.450) GO TO TOO
ITCTAL=ITCTAL+1
ICEIG+1
ICEIG+1
ITYPE(I)=1
RTMAX(I)=.65
RTMIN(I)=.11
ITYPE(I)=1
RTMAX(I)=.55
RTMIN(I)=.11
ITYPE(I)=1
RTMAX(I)=.55
RTMIN(I)=.11
ITYPE(I)=1
RTMAX(I)=.55
RTMIN(I)=.11
ITYPE(I)=1
RTMAX(I)=.45
RTMAX(I)=.45
RTMAX(I)=.45
RTMAX(I)=.45
RTMAX(I)=.11
ITYPE(I)=1
RTMAX(I)=1.
ITYPE(I)=1
              755
                 756
              757
              758
              759
790
                                                                                                                                                                                                                                                                                                                                                                                                        RANDOMIZE SPECTRUM A LOACS FOR STMIN
```

```
ITCTAL=0
IX=583
CALL RANDU(IX,IY,YFL)
                                                                                                                                                                                         | X=583
| CALL RANDU(IX,IY,YFL)
| IX=IY
| IF(ITOTAL.EQ.101) GD TO 791
| IF(IYFL.GE.0.1.AND.(YFL.LT..06)) GI
| IF(IYFL.GE..07).AND.(YFL.LT..20) GI
| IF(IYFL.GE..21).AND.(YFL.LT..27) GI
| IF(IYFL.GE..28).AND.(YFL.LT..34) GI
| IF(IYFL.GE..35).AND.(YFL.LT..34) GI
| IF(IYFL.GE..35).AND.(YFL.LT..48) GI
| IF(IYFL.GE..49).AND.(YFL.LT..48) GI
| IF(IYFL.GE..56).AND.(YFL.LT..62) GI
| IF(IYFL.GE..63).AND.(YFL.LT..62) GI
| IF(IYFL.GE..84).AND.(YFL.LT..83) GI
| IF(IX.EC.50) GO TO 773 GI
| ITOTAL = ITOTAL+1 GI
| IA=IA+1 GI
| 
   773
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           GO TO 760

1 GO TO 761

2 GO TO 763

3 GO TO 764

3 GO TO 765

4 GO TO 766

4 GO TO 766

4 GO TO 768

4 GO TO 769

4 GO TO 770

5 GO TO 771

6 GO TO 771

6 GO TO 771

6 GO TO 773
   760
761
                                                                                                                                                                                             IB=IB+1

HCLD(I)=-.04

I=I+1

GC TO 773

IF(IC.EC.10) GO TO 773

ITCTAL=ITOTAL+1

IC=IC+1

+CLC(I)=-.08

I=I+1

GC TO 773

IF(ID.EC.6) GO TO 773

ITCTAL=ITCTAL+1

HCLC(I)=-.12

I=I+1

HCLC(I)=-.12

I=I+1

HCLC(I)=-.12

I=I+1

HCLD(I)=-.16

I=I+1

HCLD(I)=-.16

I=I+1

HCLD(I)=-.16

I=I+1

HCLD(I)=-.16

I=I+1

HCLD(I)=-.16

I=I+1

HCLD(I)=-.20

I=I+1

HCLD(I)=-.20

I=I+1

HCLD(I)=-.20

I=I+1

HCLD(I)=-.20

I=I+1

HCLD(I)=-.20
762
   763
   764
   765
                                                                                                                                                                                             HCLD(I) =- .20
I = I+1
GC TO 773
IF (IG. EG. 3) GO TO 773
ITCTAL = ITGTAL +1
IG = IG+1
HCLD(I) =- .25
I = I+1
GC TO 773
IF (IH. EG. 2) GO TO 773
ITCTAL = ITCTAL +1
IF = IH+1
IF
   766
   767
   768
```

```
IF(IJ.EC.1) GO TO 773
ITCTAL=ITGTAL+1
IJ=IJ+1
HCLD(I)=-.37
I=I+1
GC TO 773
IF(IK.EQ.1) GO TO 773
ITCTAL=ITOTAL+1
IK=IK+1
HCLC(I)=-.41
I=I+1
IGC TO 773
IF(IL.EQ.1) GO TO 773
ITCTAL=ITGTAL+1
IL=IL+1
HCLC(I)=-.45
I=I+1
GO TO 773
CCNTINUE
 769
 770
 771
791
                    RANCOMIZE SPECTRUM A STMIN LOADS WITH .11LL(1G.) LCACS TO EQUAL NUMBER OF STMAX LOADS
                   IA=0

IE=0

J=1

I=1

ITGTAL=0

IX=4777

CALL RANDU(IX,IY,YFL)
 780
                   IX=IY
IX=IY
IF(ITOTAL.EQ.4201) GO TO 792
IF((YFL.GE.0.).AND.(YFL.LT..2)) GC TO 781
IF((YFL.GE.0.2).ANC.(YFL.LT..1.)) GC TO 782
IF(IA.EQ.101) GO TO 780
ITCTAL=ITCTAL+1
IA=IA+1
RTMIN(J)=HCLD(I)
 781
                   J=J+1
J=J+1
GC TO 780
IF(IB.EC.4100) GO TO 780
ITCTAL=ITCTAL+1
IB=IB+1
RTMIN(J)=.11
J=J+1
GC TO 780
CCNTINUE
 782
792
C
C
C
C
                    ESTABLISH A GROUND CYCLE EVERY HOUR ( 42 EVENTS)
                   CC 795 J=1,4201,42
RTMIN(J)=-.08
CCNTINUE
```

## LIST OF REFERENCES

- 1. Coffin, L. F., Jr., "Low Cycle Fatigue: A Review,"
  Applied Materials Research, v. 1, p. 129-141, October 1972.
- Vought Aeronautics Division Report No. 2-53420/OR-5562, <u>A7 Wing Fatigue Life</u>, by J. W. Darnell, Jr. and J. W. <u>Cherry</u>, p. 1.17, 6 August 1970.
- 3. Department of Defense Specification MIL-HDBK-5A, Metallic Materials and Elements for Aerospace Vehicles Structures, February 1966.
- 4. Dowling, N. E., "Fatigue Failure Predictions for Complicated Stress-Strain Histories," <u>Journal of Materials</u>, v. 7, p. 71-87, March 1972.
- 5. Manson, S. S. and Hirschberg, M. H., <u>Fatigue Behavior in Strain Cycling in the Low- and Intermediate-Cycle Range</u>, paper presented at Sagamore Army Materials Research Conference, 10th, Raquette Lake, New York, 13 August 1963.
- 6. Air Force Flight Dynamics Laboratory Technical Report AFFDL-TR-74-23, A User's Manual for the Sequence Accountable Fatigue Analysis Computer Program, by J. M. Potter and R. A. Noble, May 1974.
- Sandor, B. I., Fundamentals of Cyclic Stress and Strain, p. 64-84, University of Wisconsin Press, 1972.
- 8. Agardograph No. 157, The Accumulation of Fatigue Damage in Aircraft Materials and Structures, by J. Schijve, January 1972.
- 9. Tavernelli, J. F. and Coffin, L. F., Jr., "Experimental Support for Generalized Equation Predicting Low Cycle Fatigue," <u>Journal of Basic Engineering</u>, v. 84, p. 533-541, December 1962.
- 10. van Dijk, G. M., Statistical Load Data Processing, paper presented at International Committee on Aeronautical Fatigue (ICAF) Symposium on Advanced Approaches to Fatigue Evaluation, 6th, Miami Beach, Florida, 13 May 1971.
- 11. Wetzel, R. M., "Smooth Specimen Simulation of Fatigue Behavior of Notches," <u>Journal of Materials</u>, v. 3, p. 646-657, September 1968.

## INITIAL DISTRIBUTION LIST

		No. Copies
1.	Defense Documentation Center Cameron Station	2
	Alexandria, Virginia 22314	
2.		2
	Naval Postgraduate School Monterey, California 93940	
3.	Department Chairman, Code 67 Department of Aeronautics	1
	Naval Postgraduate School Monterey, California 93940	
4.		· 1
	Department of Aeronautics Naval Postgraduate School	
	Monterey, California 93940	
5.	Lieutenant John Scott Atkinson, Jr., USN 221 West Street	1.
	Winchester, Virginia 22601	